

DEVELOPMENT OF THE CONTROLLABLE TRAILING EDGE FLAP TECHNOLOGY FOR MW TURBINES

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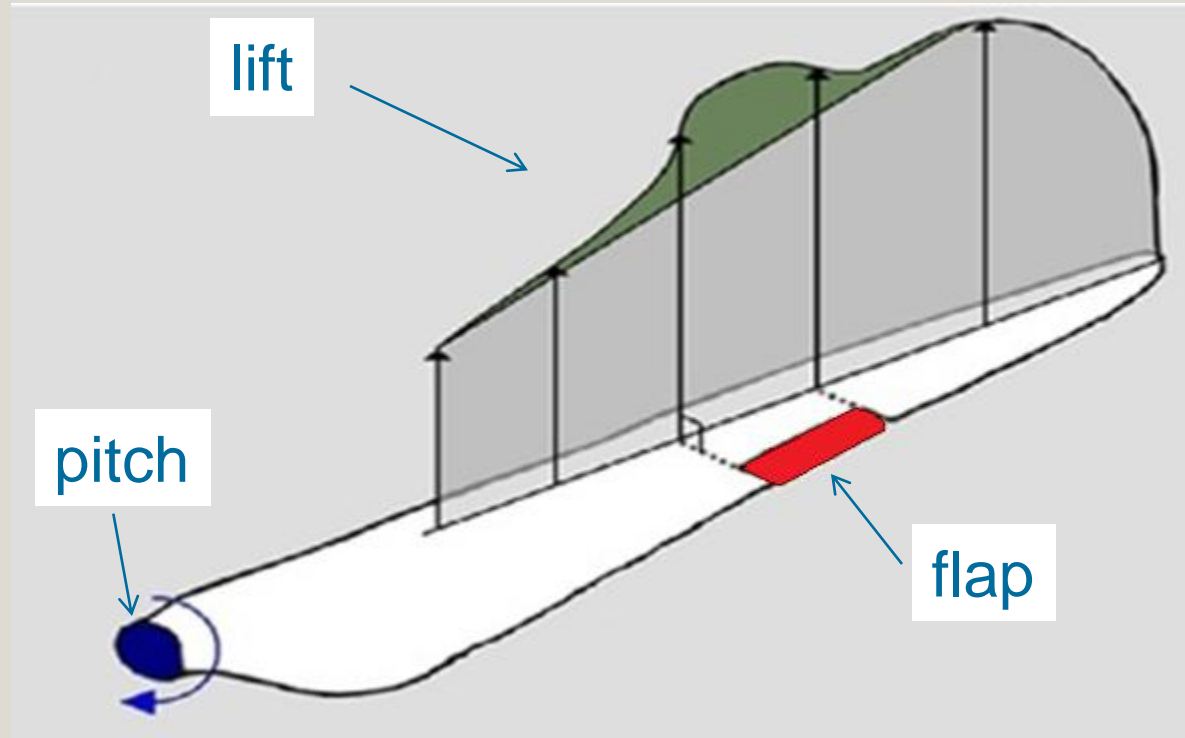
AGENDA

Load alleviation potential using trailing edge flaps

Status of the controllable rubber flap prototype tests

Challenges in the implementation of the flap system on MW turbines

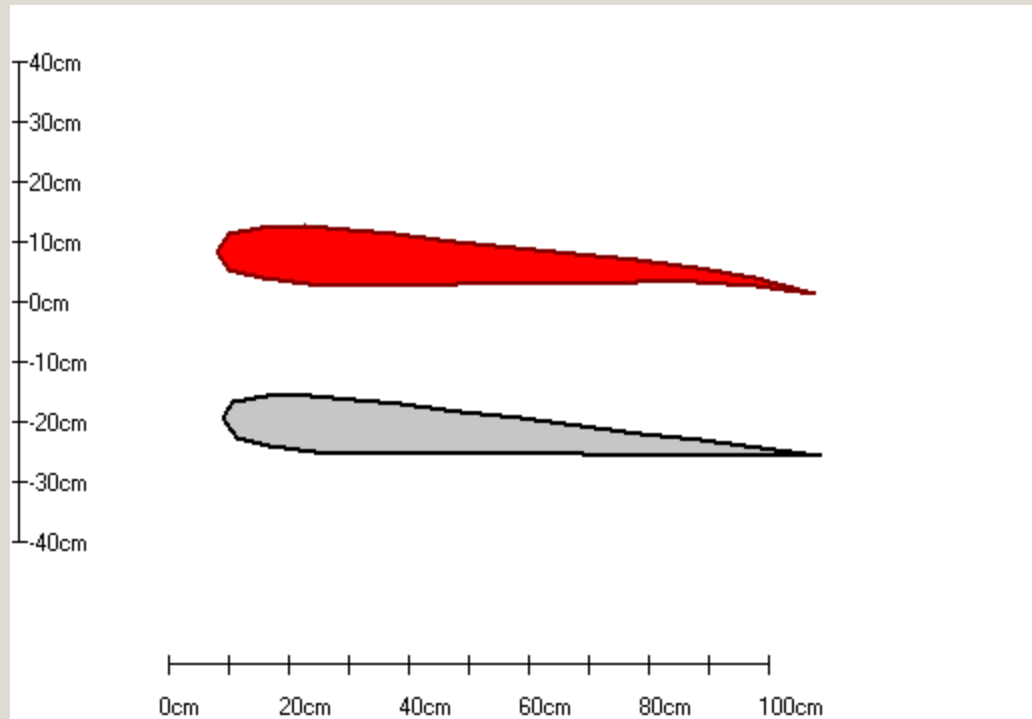
BACKGROUND (1)



Flap down (+)

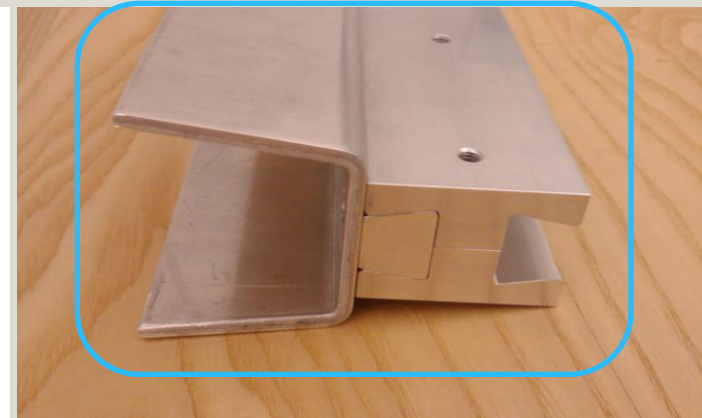
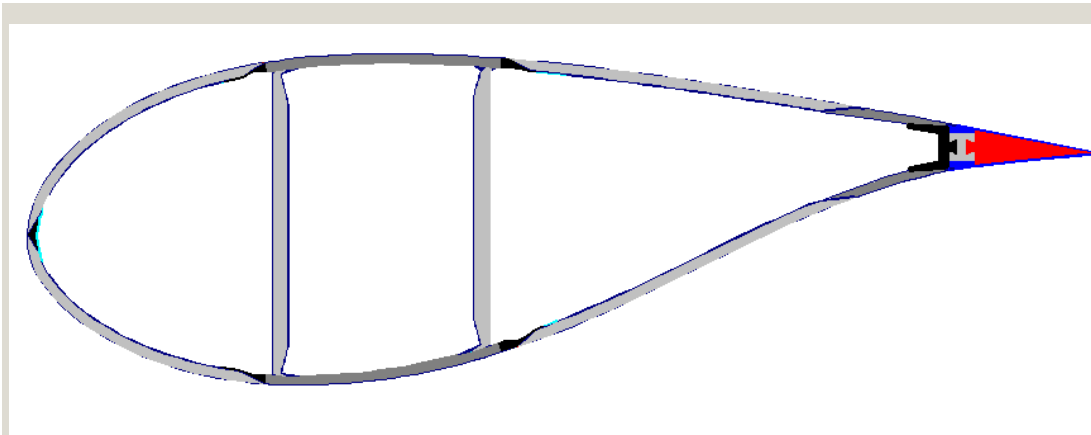
- pitch
- flap **local** load control

BACKGROUND (2)



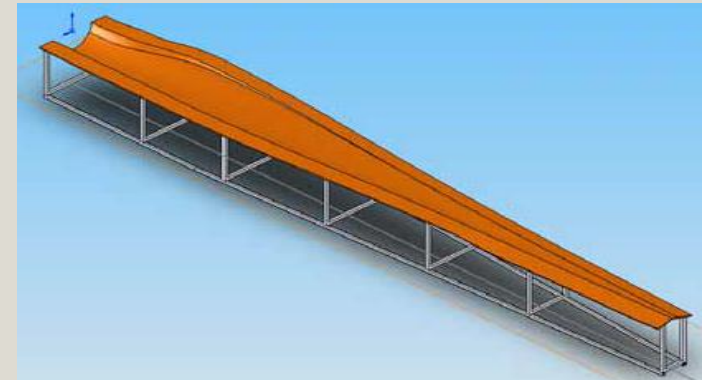
- Gray airfoil section without flap
- Red airfoil section with flap

FLAP DESIGN (1)

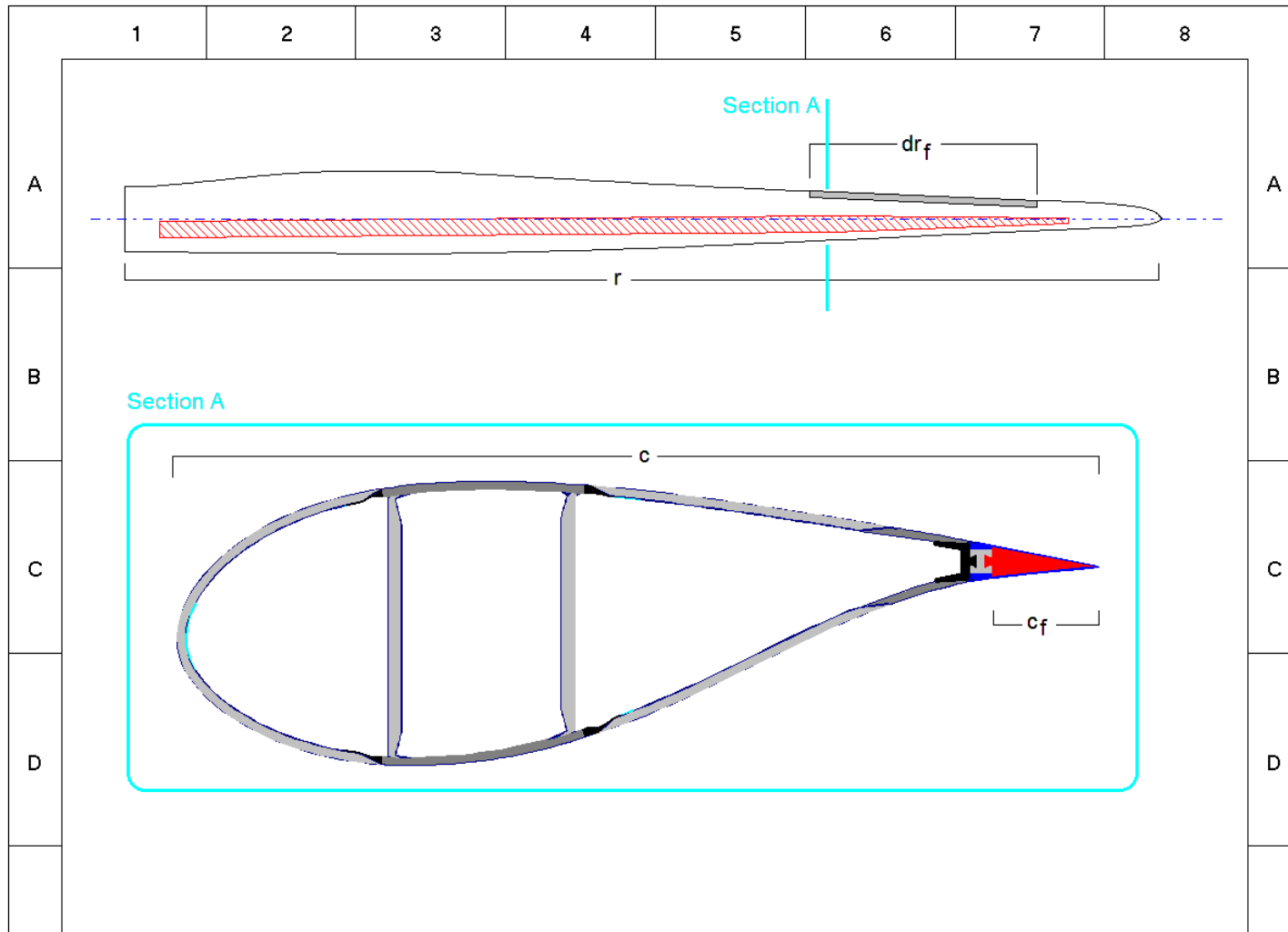


Aim to reduce mould layup time
TE thickness low - 2mm or less

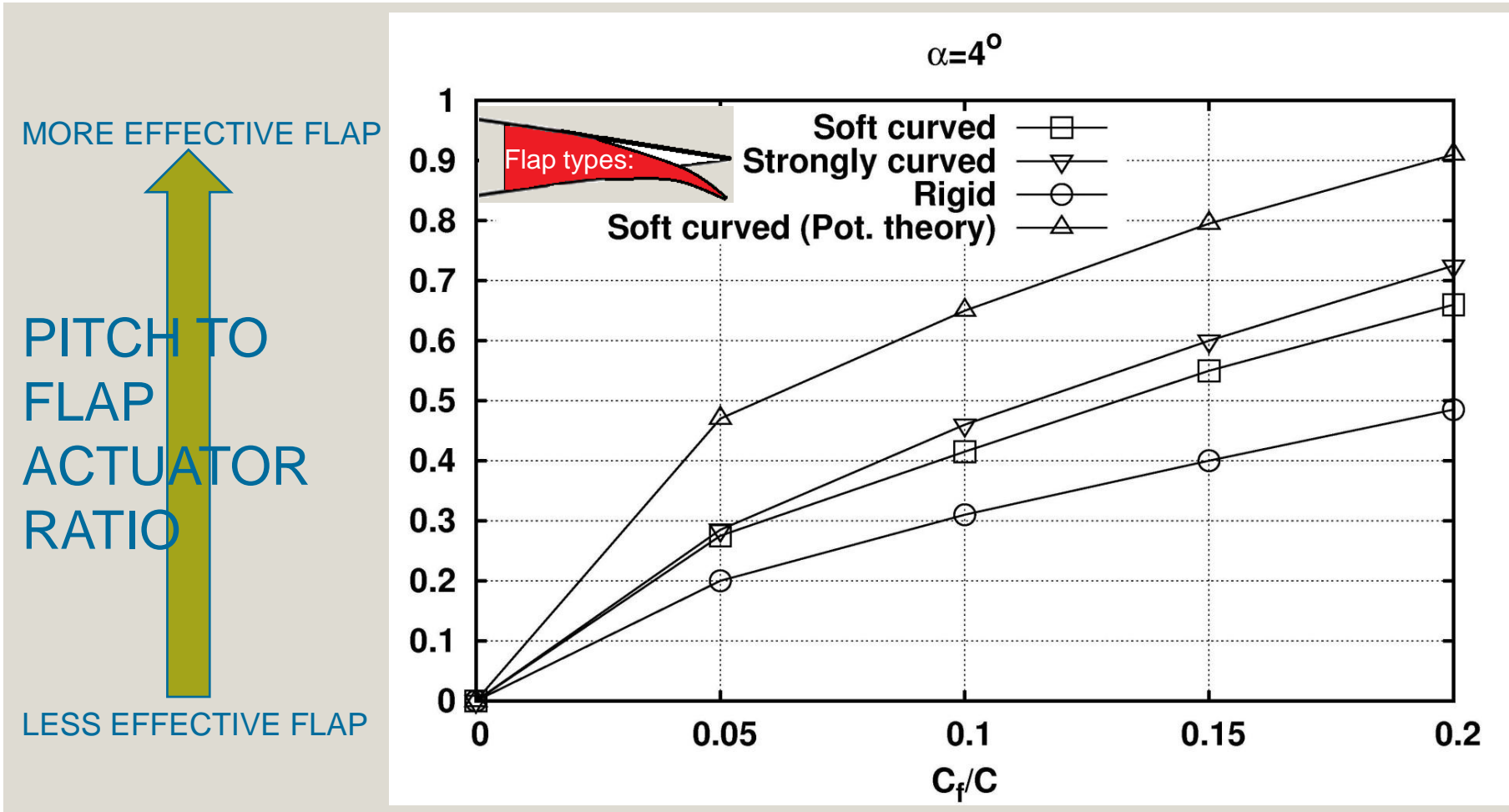
OVERALL GOAL:
NO MECHANICAL PARTS IN THE BLADE



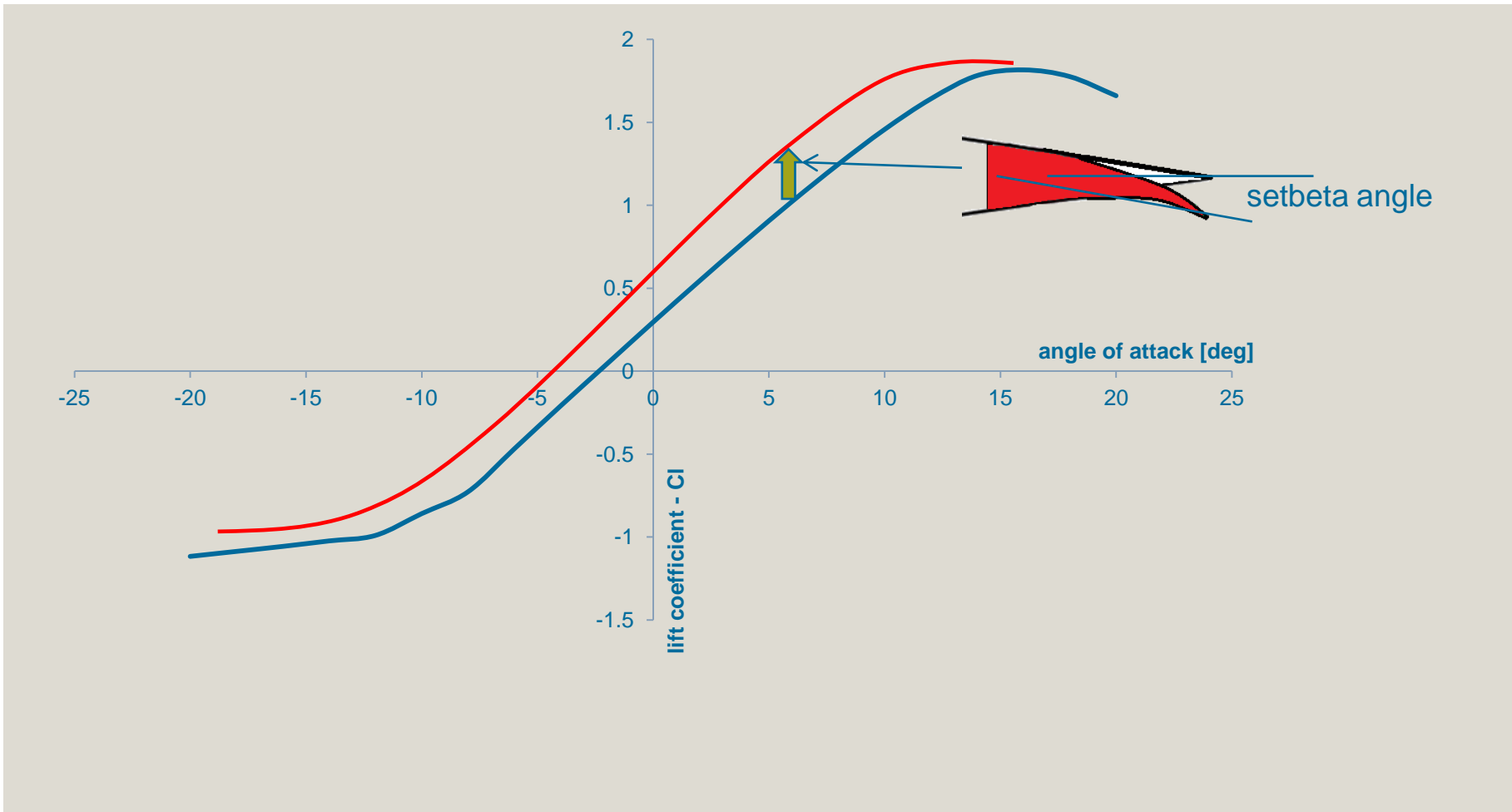
FLAP DESIGN (2)



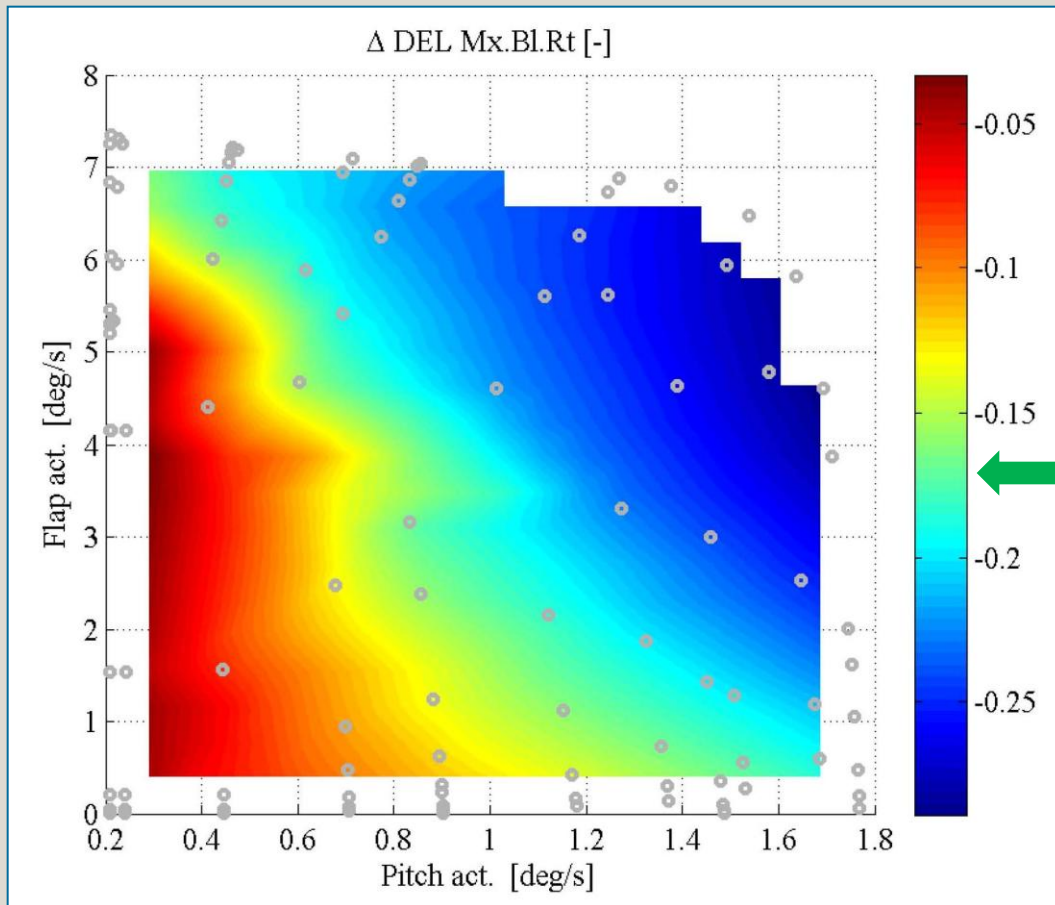
FLAP DESIGN (3)



FLAP "MUSCLE"



CONTROL BASED ON TRADITIONAL SENSORS



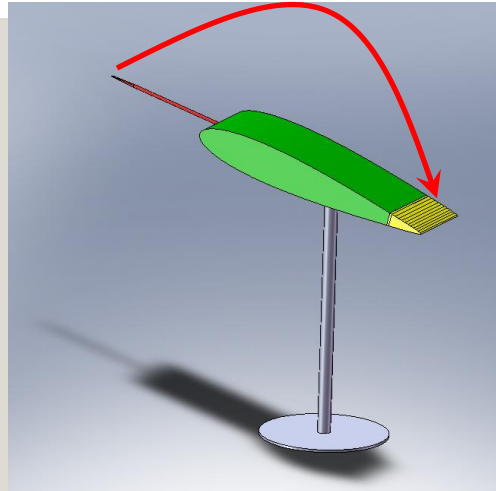
CONTROL OBJECTIVE:
Reduce the pitch activity and
alleviate the loads using the
 same sensors as for the pitch
 system

Fatigue Damage Equivalent
 Loads (DEL) alleviation at
 the blade root flapwise
 bending compared to the
 baseline NREL 5 MW
 turbine, Wöhler curve
 exponent of 10 and IEC
 class IA – dlc1.1 at 18m/s
 (4 seeds)

INFLOW SENSORS (ADVANCED)



Experiment carried out within the DAN-AERO project from 2007-2010: LM, Vestas, Siemens, DONG Energy and Risø DTU



CONTROL OBJECTIVE:
Measure the incoming disturbance and adjust the flap position to keep the aerodynamic loading constant along the blade span

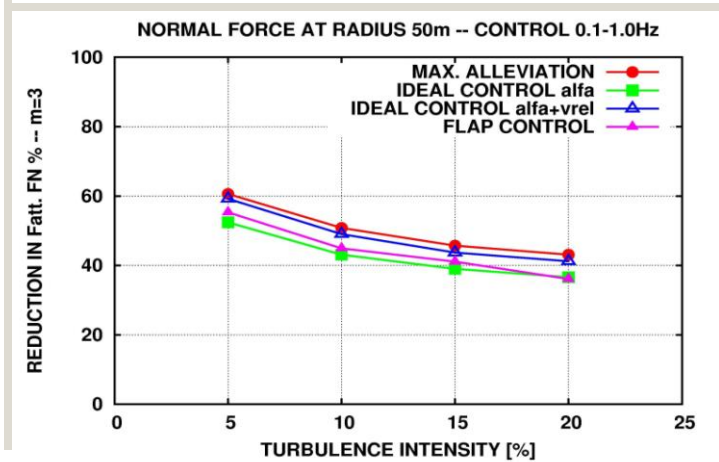
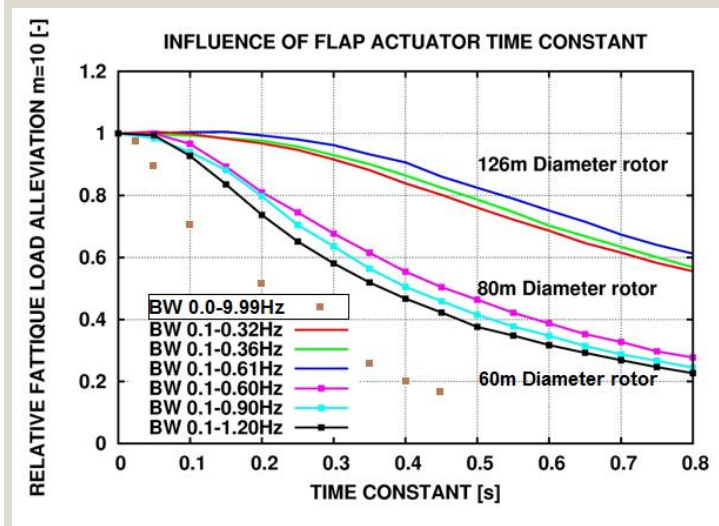


Wind tunnel test of flaps and inflow sensors, 2009: Velux windtunnel

ROTOR DIAMETER AND WIND SITE

Alleviating high frequency loads (small rotor) requires faster response time

Alleviating loads in high turbulence sites requires more “flap muscle”



EQUIVALENT FATIGUE LOAD REDUCTIONS

Table III. Comparison of results from aeroservoelastic investigations with active flaps on the Upwind 5MW RWT.

article	c_f [%]	dr_f/r [%]	δ [\pm°]	T.I. [%]	shear exp. [-]	V_{av} [m/s]	reduction in std of RBM [%]	reduction in DEL [%]	controller
Riziotis et al. 2008	10	15-47	6	-	0.2	8, 12, 16	30-35 (range)	-	PID
Andersen et al. 2008	10	63	8	14-18	0.14	7, 11, 18	-	36.2-47.9	HPF+inflow
Lackner et al. 2009	10	20	10	NTM, ETM	0.2	8, 12, 16, 20	-	5.6-24.6	PID
Barlas et al. 2009	10	20	10	NTM	0.2	8, 11.4, 16	5.7-22.4	-	PID
Andersen et al. 2009	10	15-30	8	-	11.4	-	-	25-37	HPF
Resor et al. 2010	10	24	10	6	0.2	15	26-30.9	27-31.3	PD, HPF+notch
Wilson et al. 2010	10	24	10	6	0.2	15	13.3	15.5	LQR
Berg et al. 2010	10	25	10	6	0.2	15	8.7-18.1	10.9-17	PD, LQR
this article	10	18	8	6, NTM	0.2	7, 11.4, 15	10.9-30.7	10.9-27.3	MPC+inflow

Barlas, Thanasis; Van Der Veen, Gijs; van Kuik, Gijs; Model Predictive Control for wind turbines with distributed active flaps: Incorporating inflow signals and actuator constraints. Article first published online: 17 NOV 2011 DOI: 10.1002/we.503

UPSCALED BLADE EXAMPLE (1)

Blade+hub=89.166m (baseline)



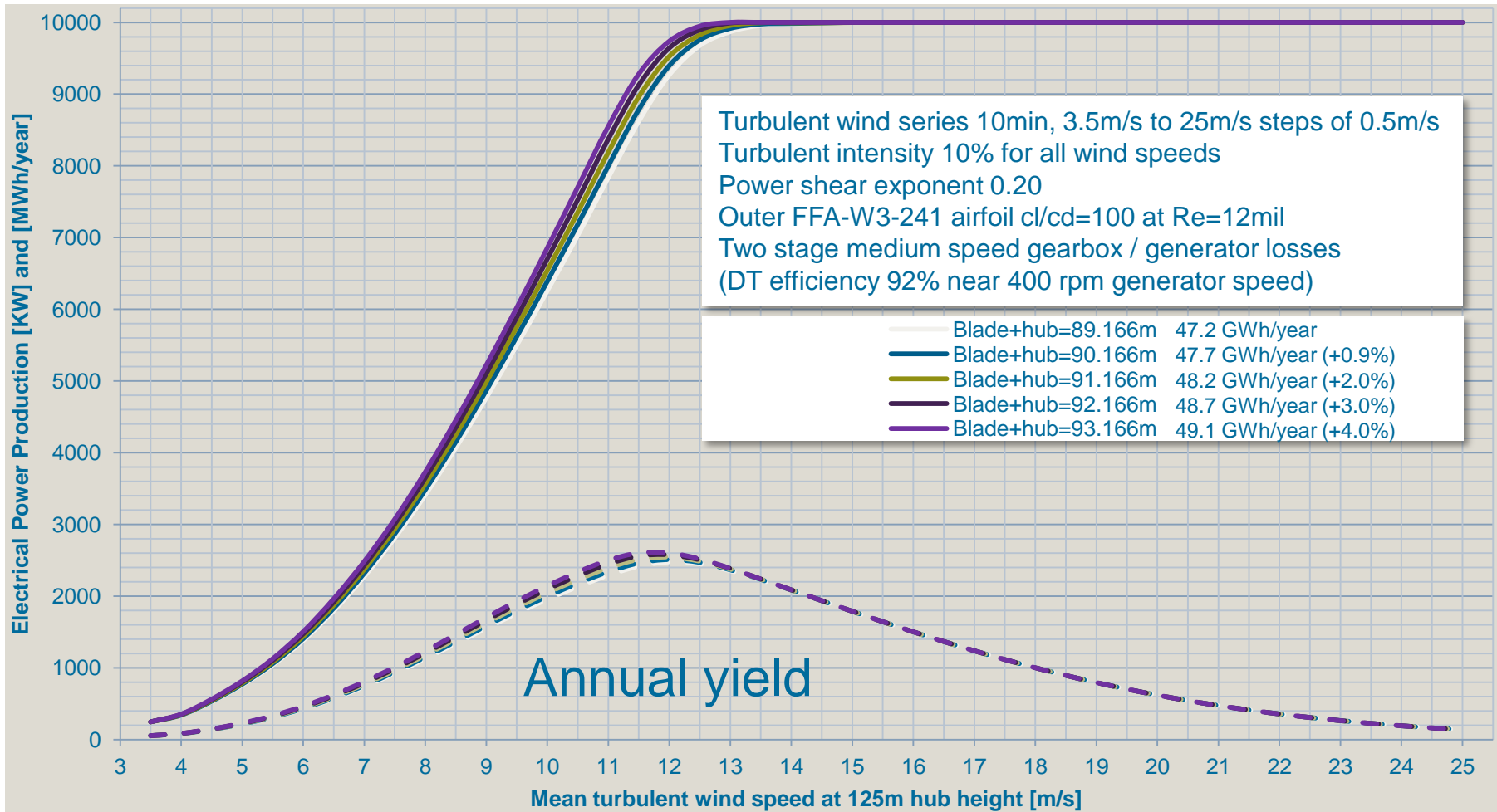
Blade+hub=90.166m

Blade+hub=91.166m

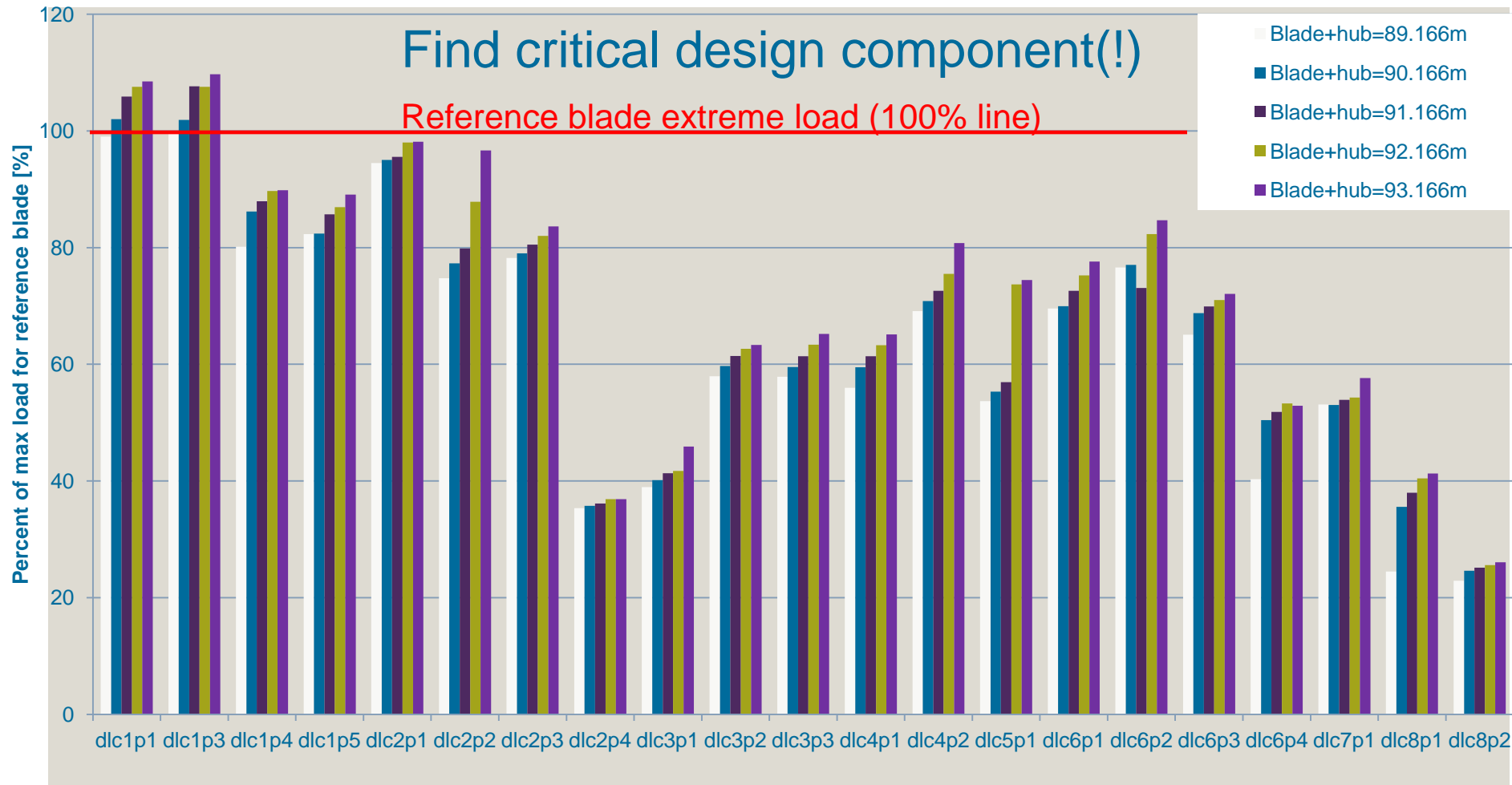
Blade+hub=92.166m

Blade+hub=93.166m

UPSCALED BLADE EXAMPLE (2)

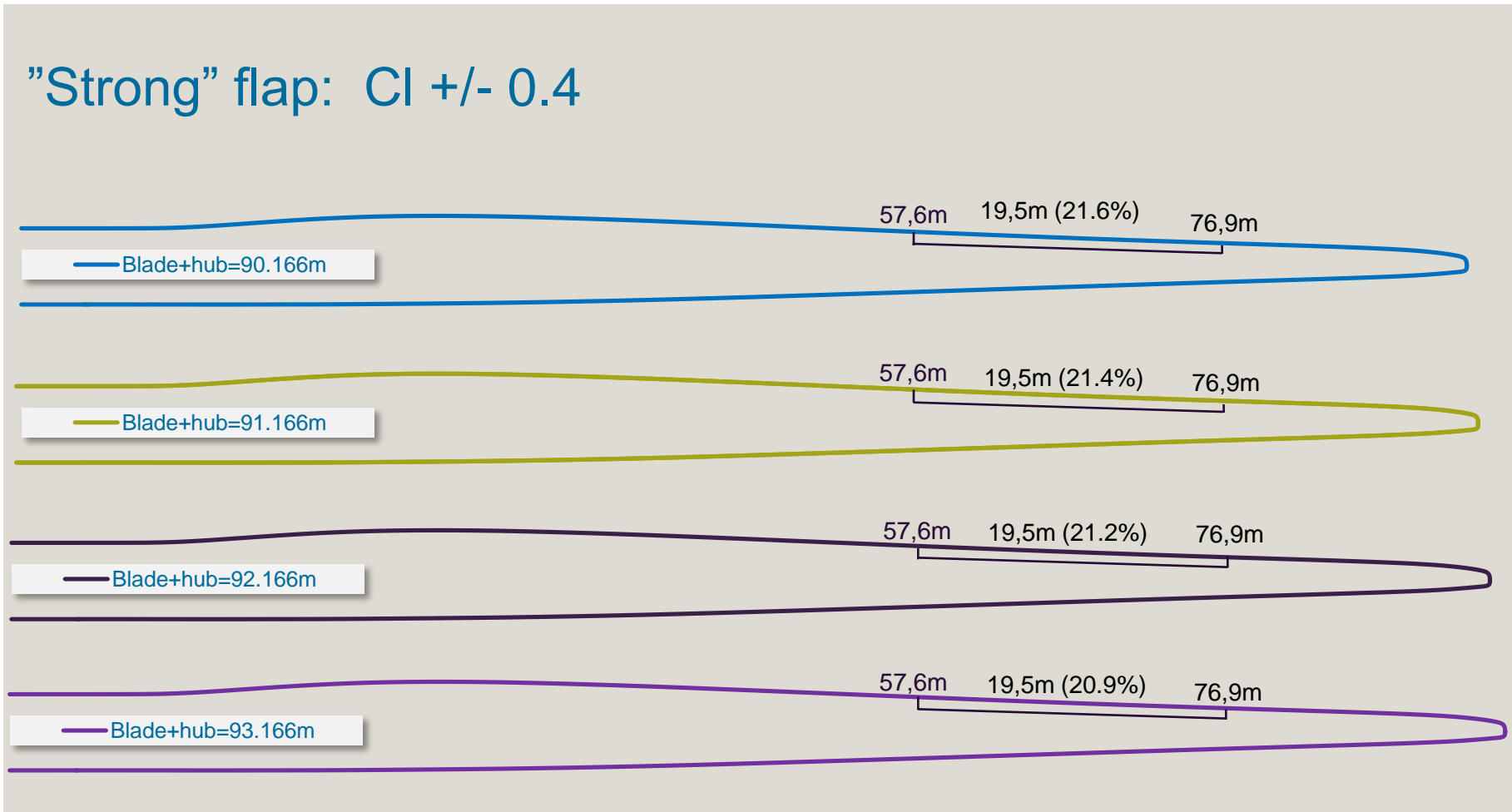


UPSCALED BLADE EXAMPLE (3)

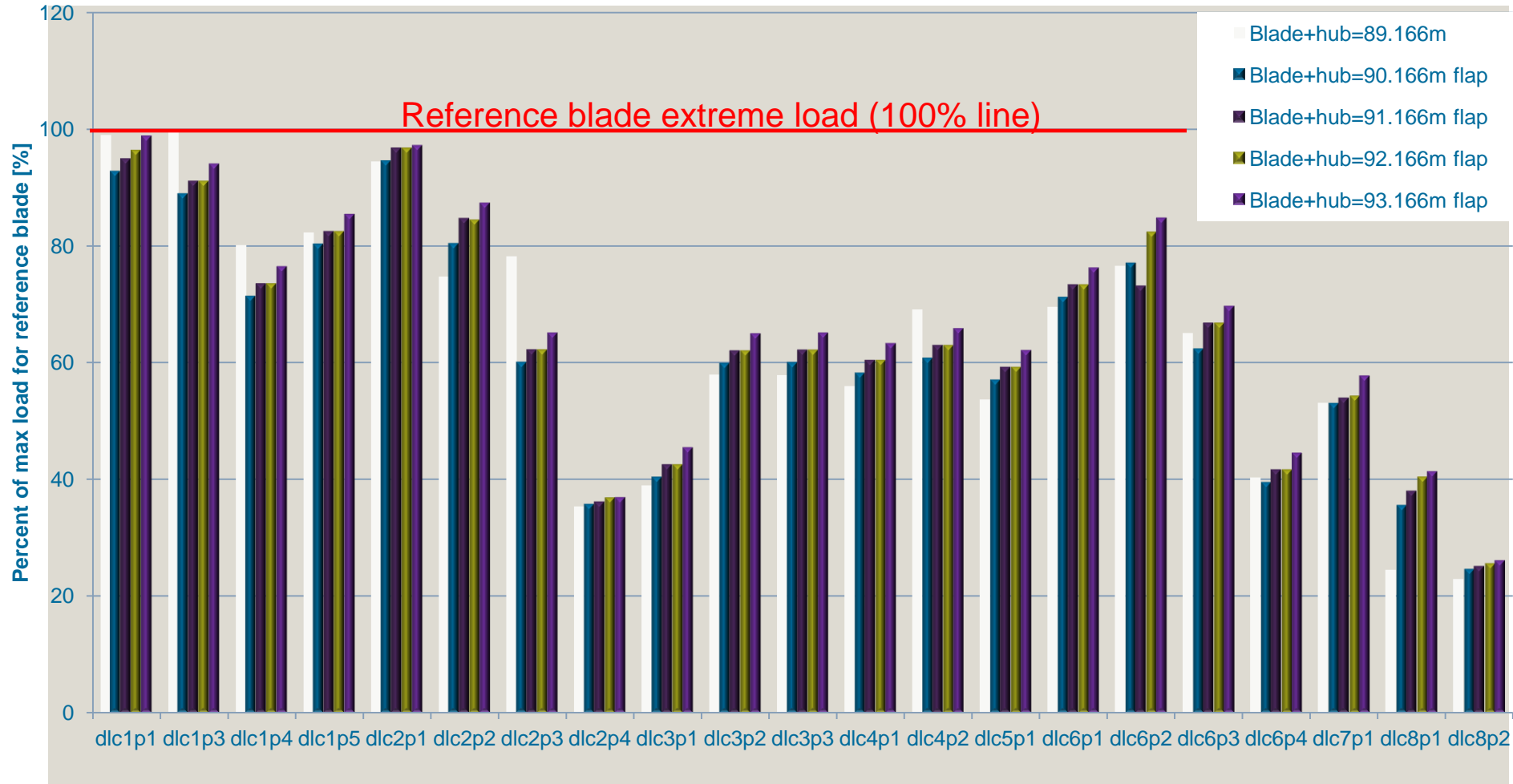


UPSCALED BLADE EXAMPLE (4)

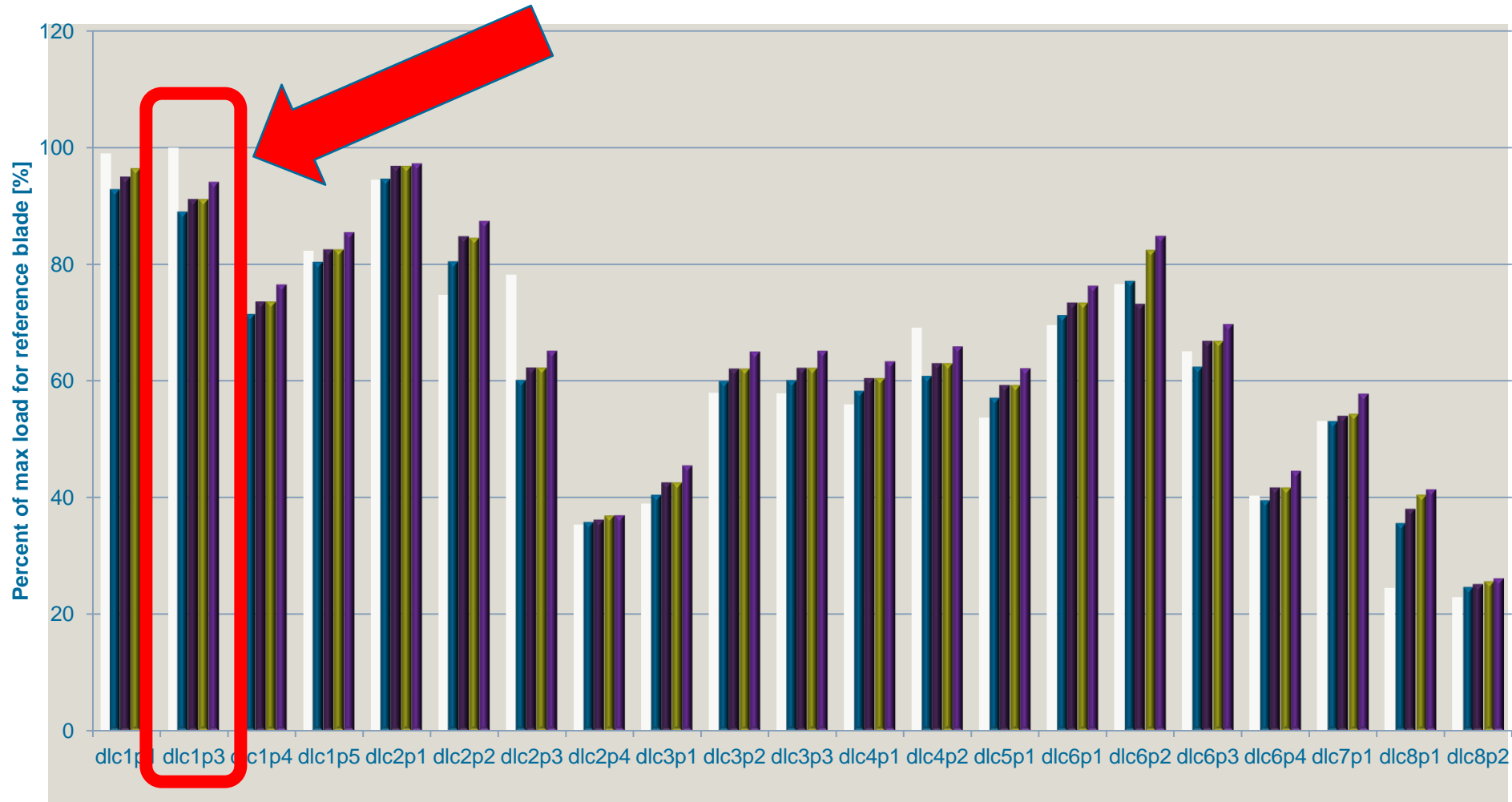
”Strong” flap: $CI \pm 0.4$



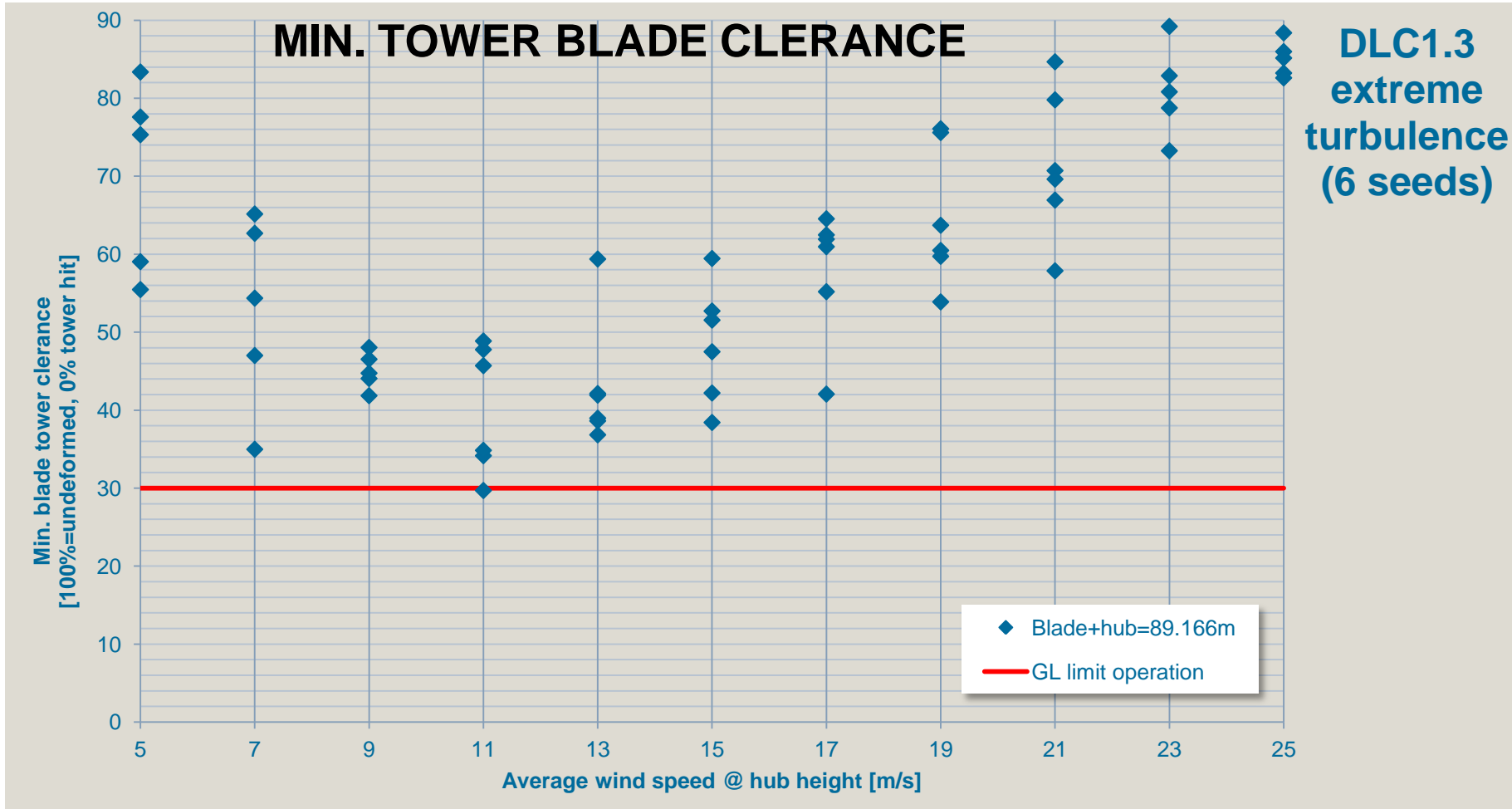
UPSCALED BLADE EXAMPLE (5)



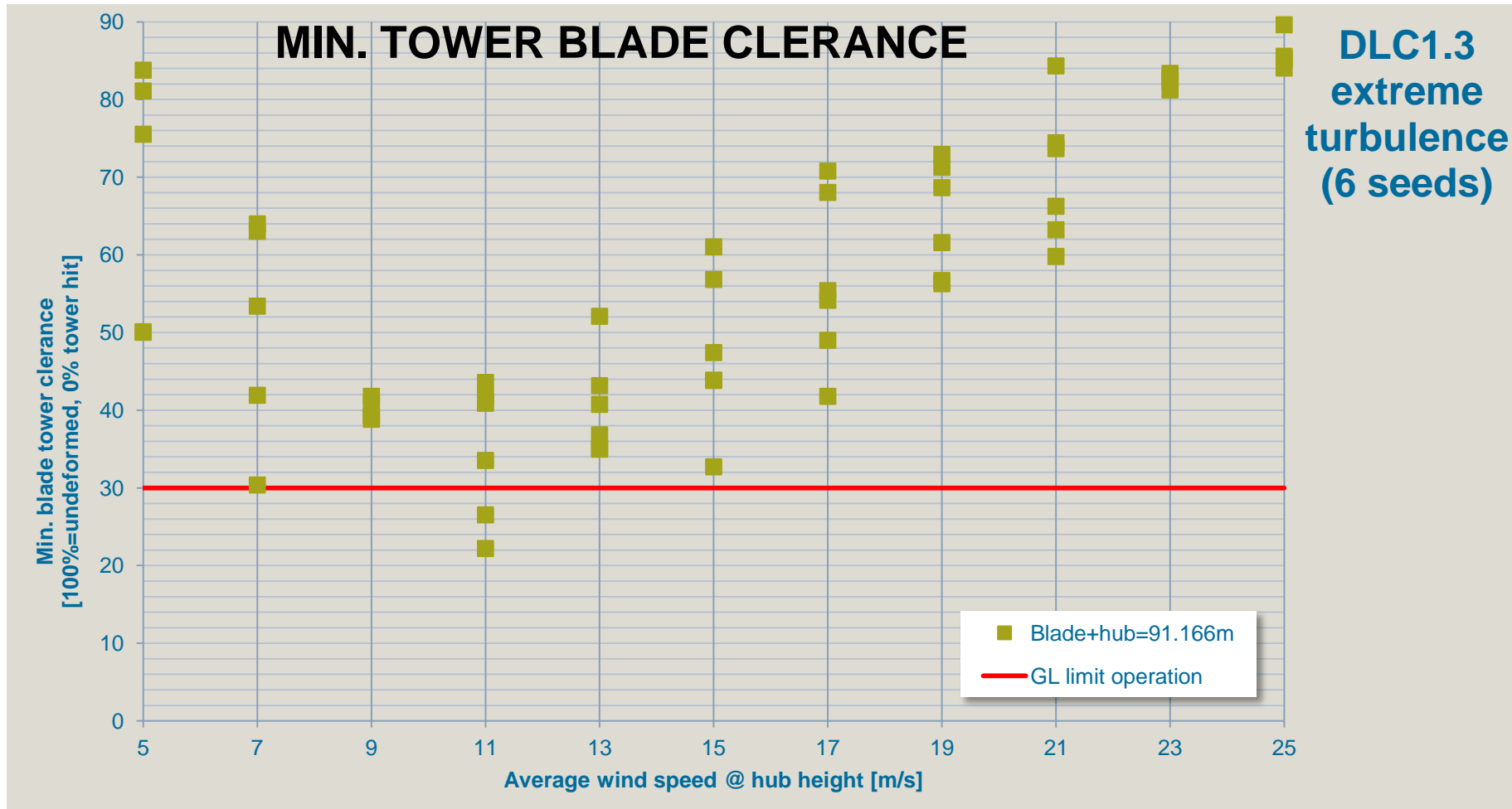
EXTREME LOADS -> TOWER-BLADE MIN. CLEARANCE



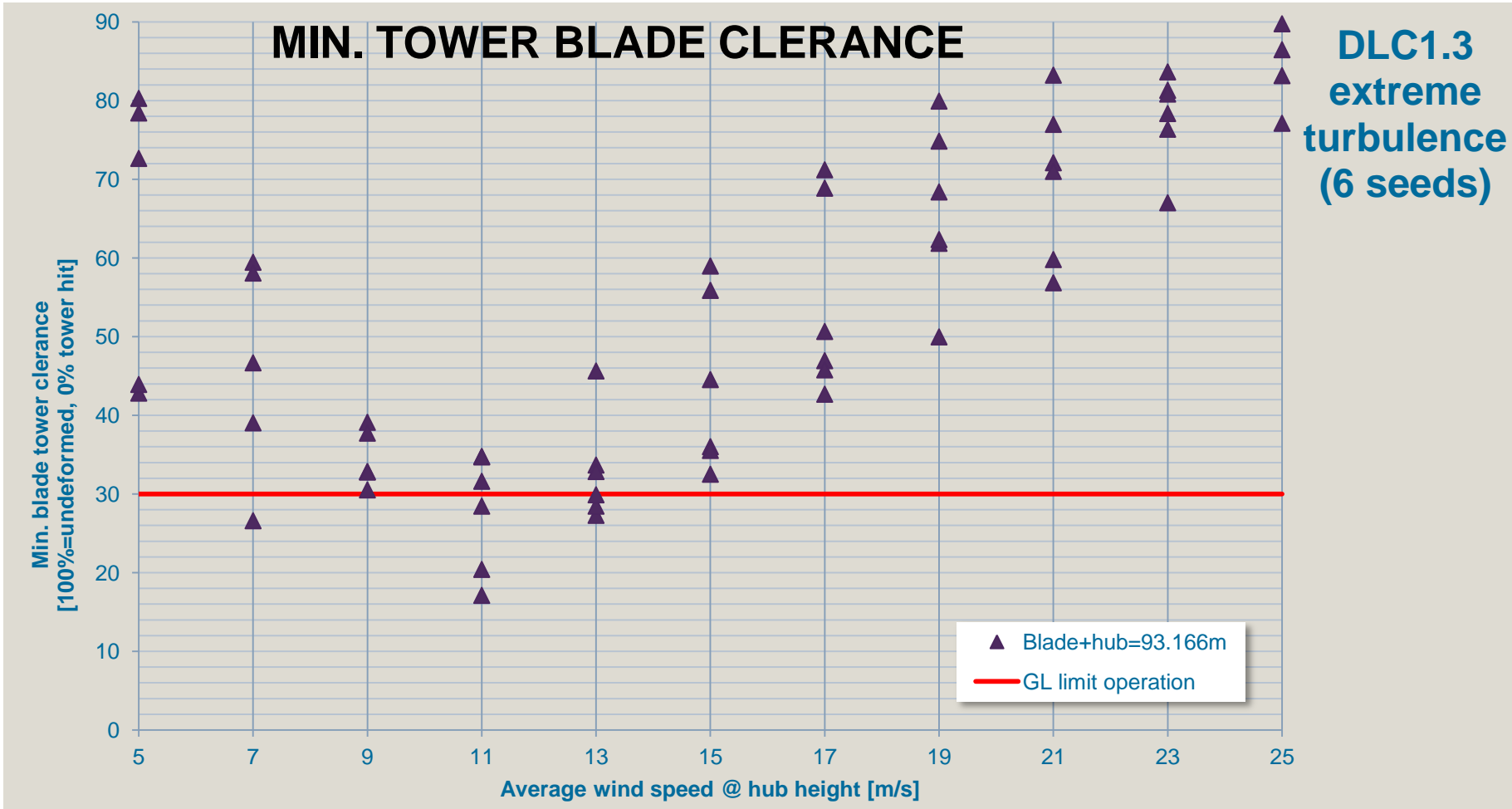
TOWER-BLADE MIN. CLEARANCE – BY GL 30% OPERATION DEMAND



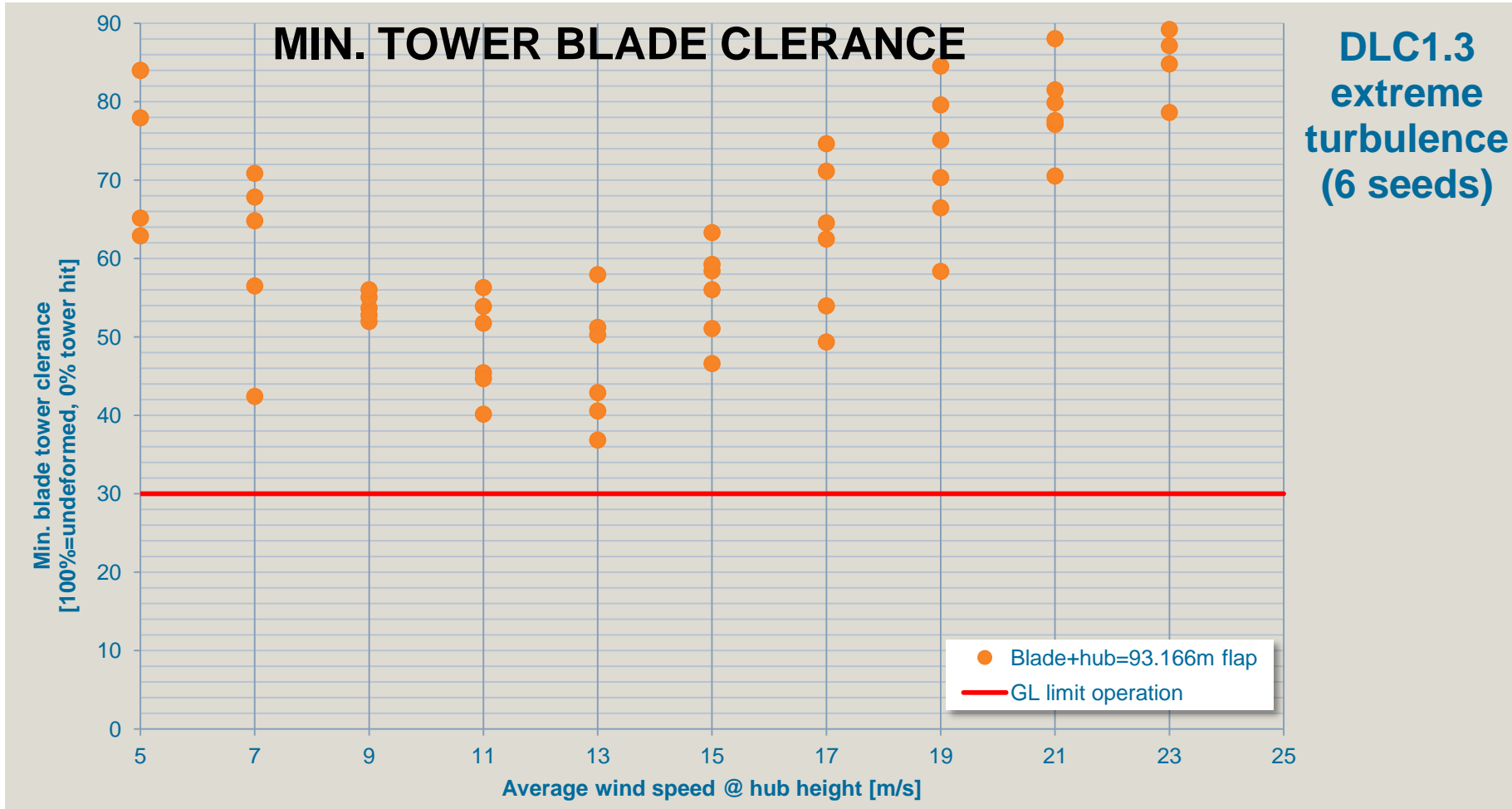
TOWER-BLADE MIN. CLEARANCE – BY GL 30% OPERATION DEMAND



TOWER-BLADE MIN. CLEARANCE – BY GL 30% OPERATION DEMAND

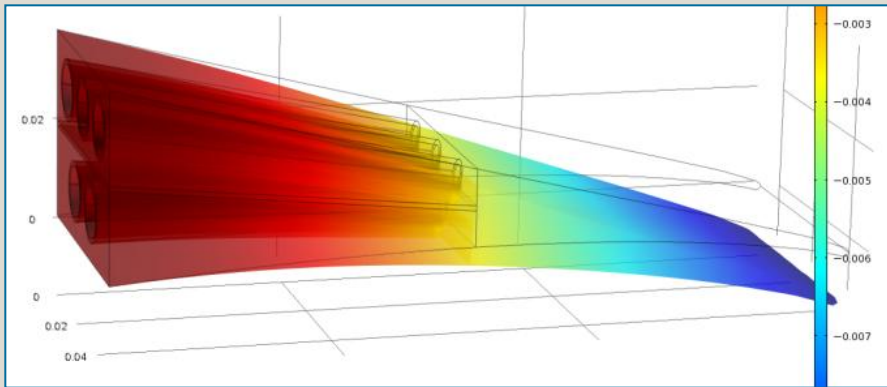


TOWER-BLADE MIN. CLEARANCE – BY GL 30% OPERATION DEMAND

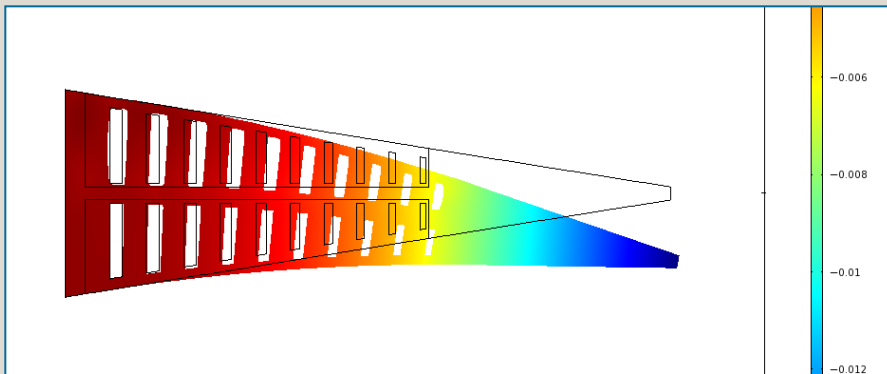


Status of the controllable rubber flap prototype tests

RUBBER FLAP (1)

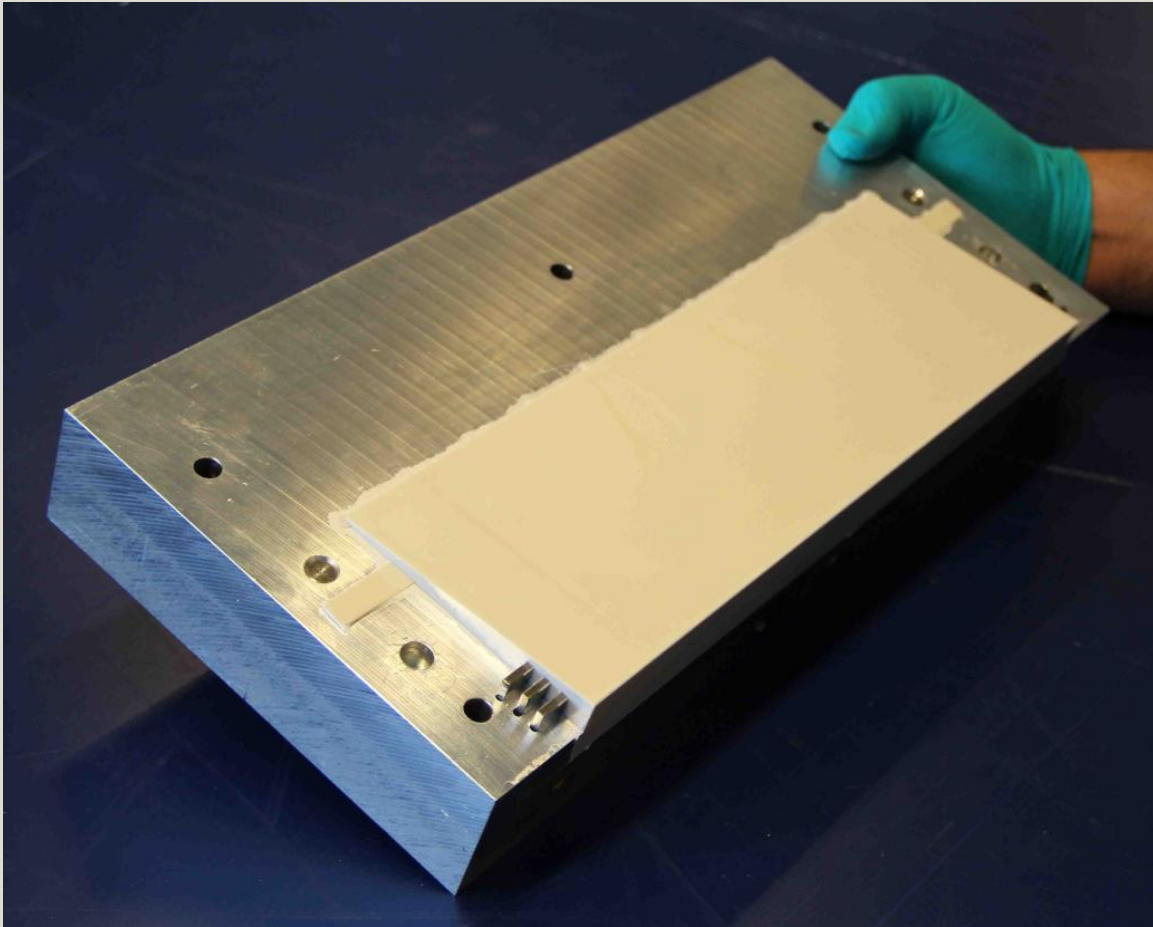


**CHORDWISE CAVITIES
FOR MOULD MANUFACTURING**



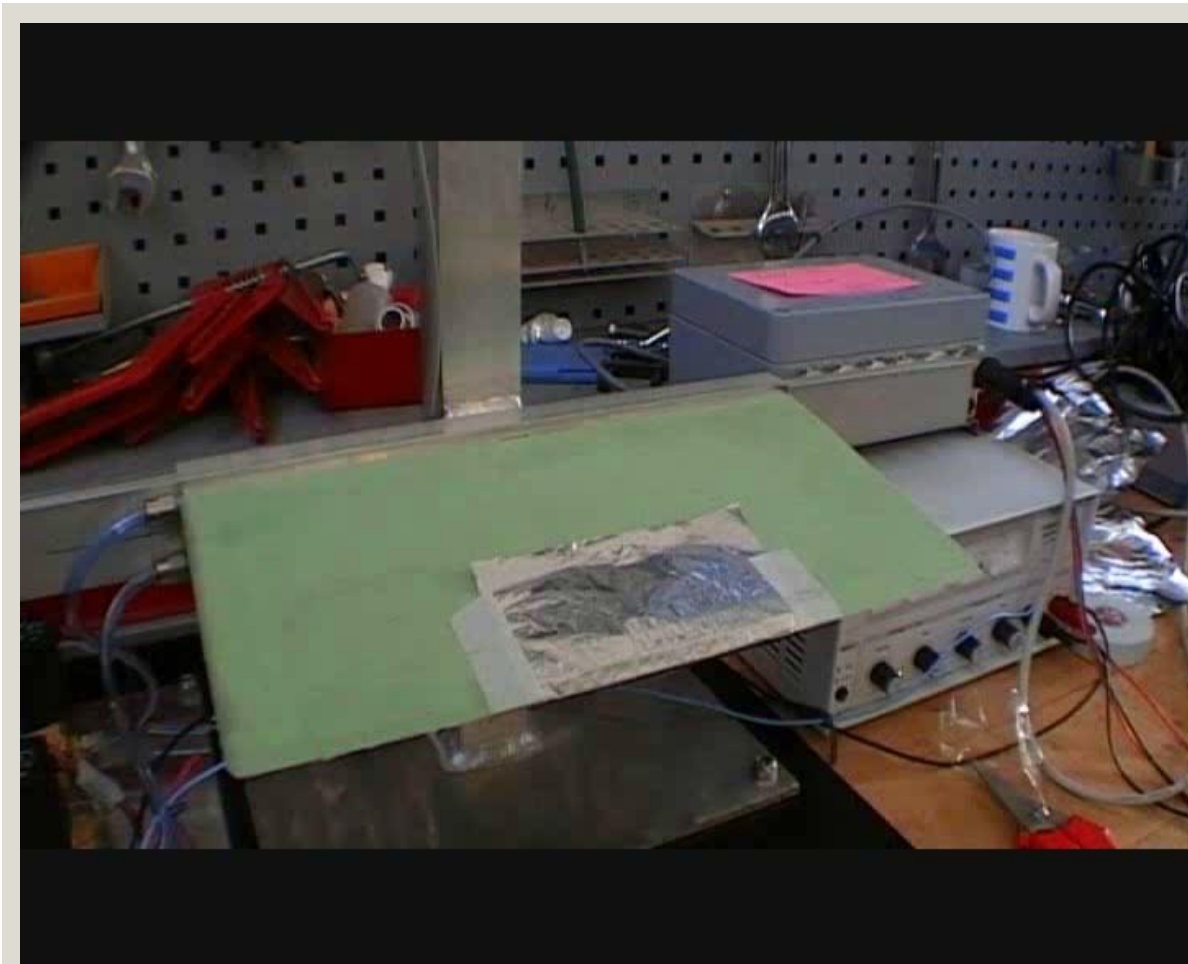
**SPANWISE CAVITIES
FOR EXTRUSION**

RUBBER FLAP (2)



PHENOMATICS
HYDROLICS

RUBBER FLAP (3)

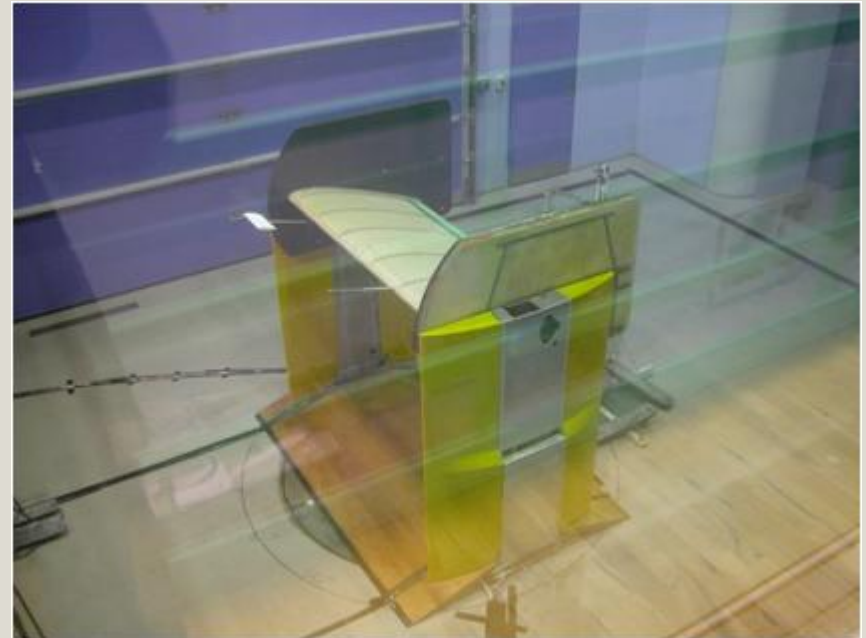


RUBBER FLAP WINDTUNNEL TEST (1)

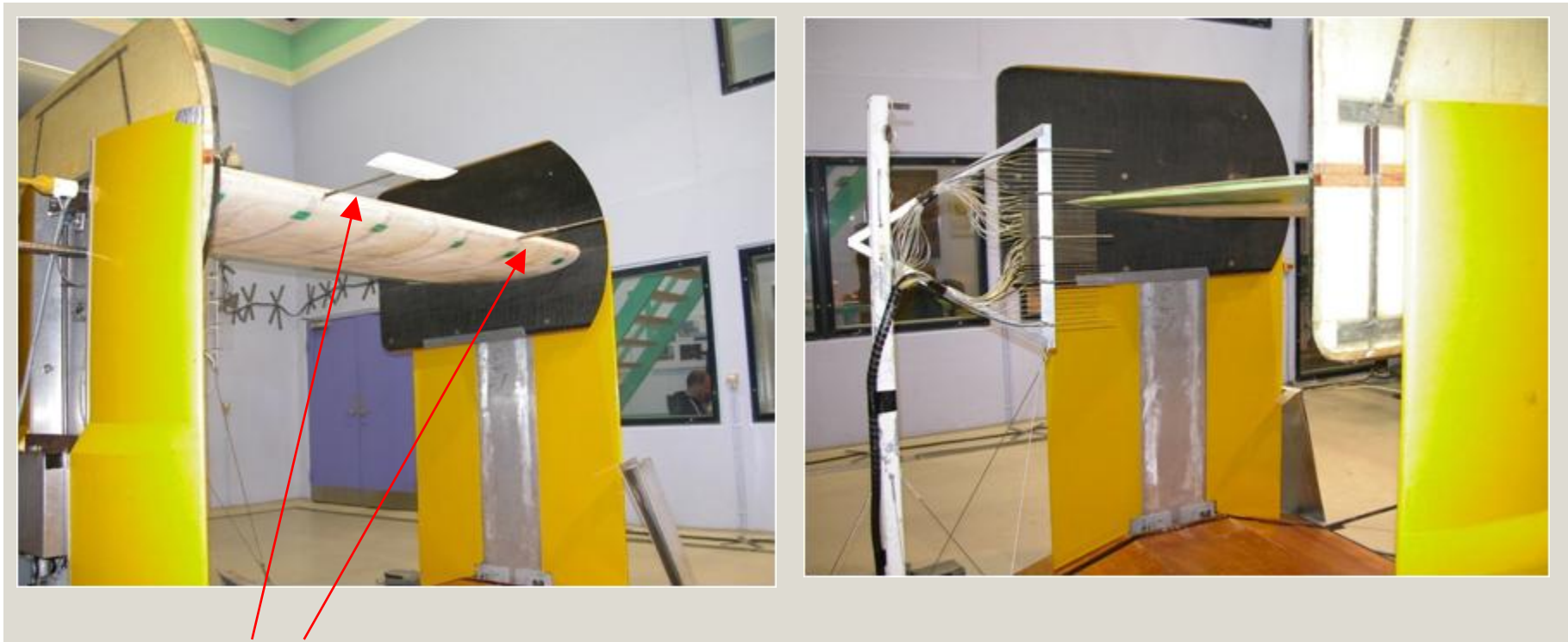
Instrumentation
2m airfoil section



VELUX wind tunnel
December 2009

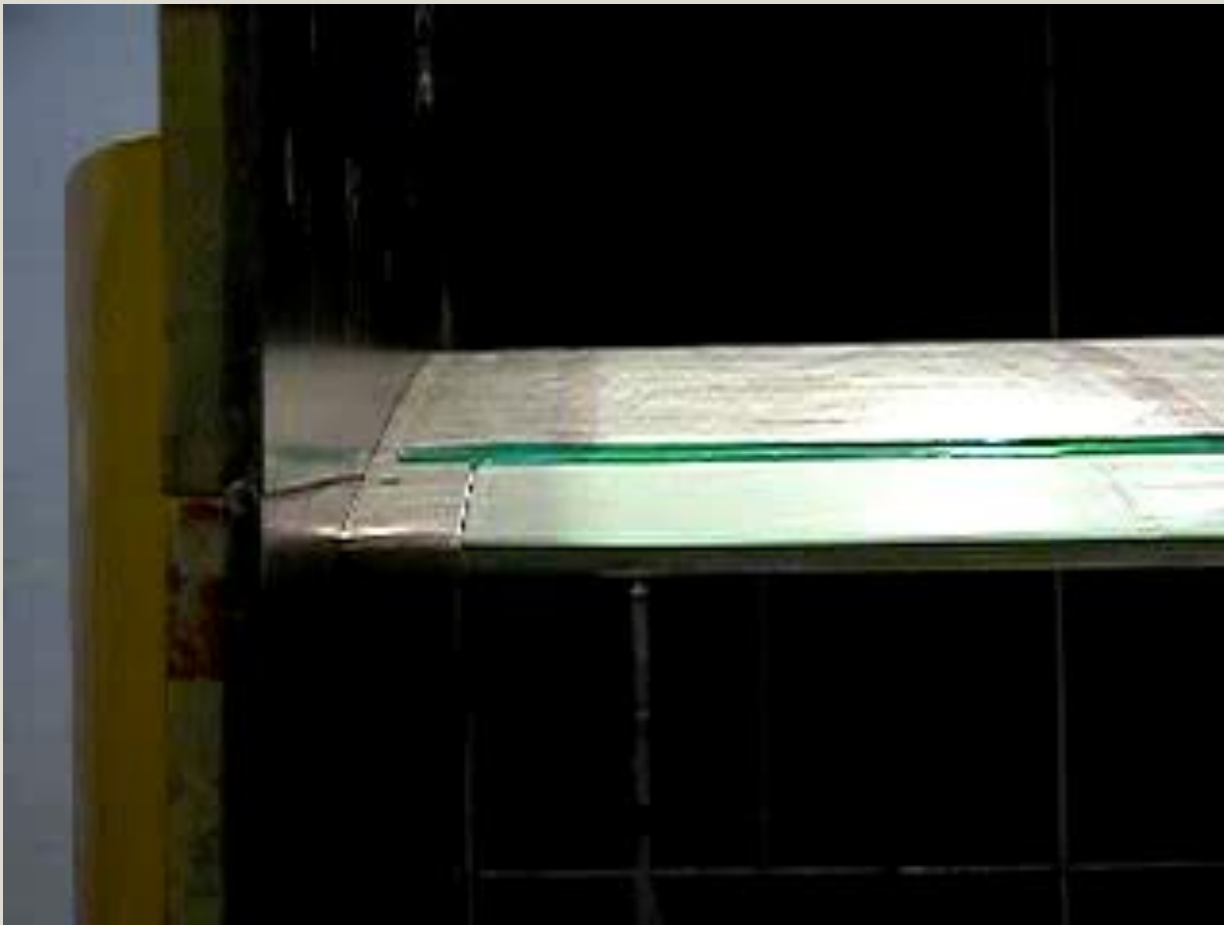


RUBBER FLAP WINDTUNNEL TEST (2)

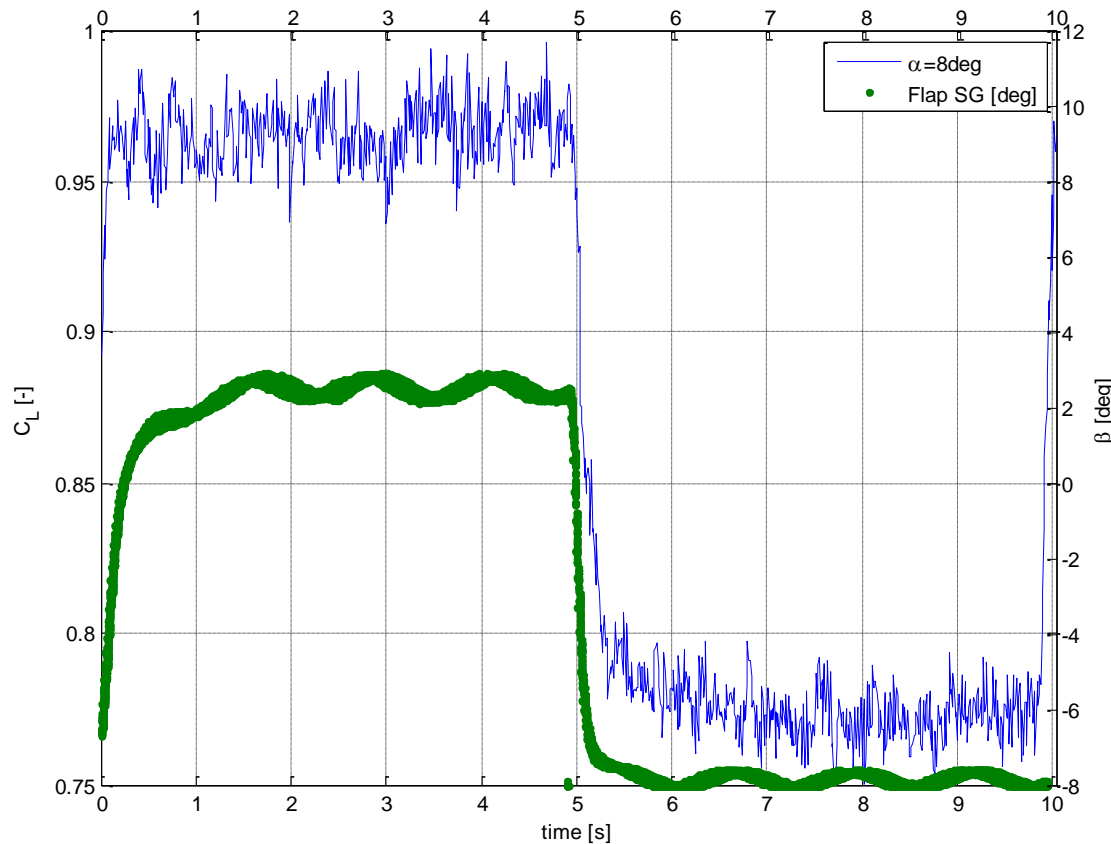


two different inflow sensors

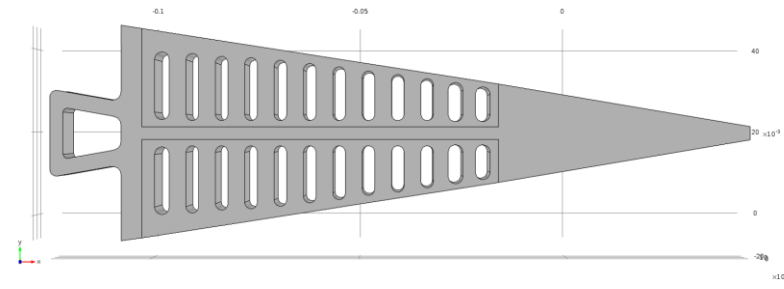
RUBBER FLAP WINDTUNNEL TEST (3)



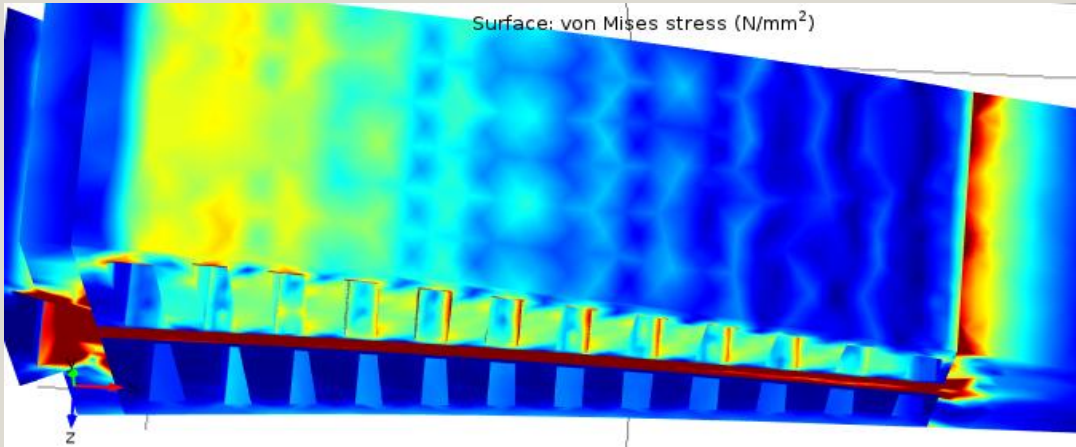
RUBBER FLAP WINDTUNNEL TEST (4)



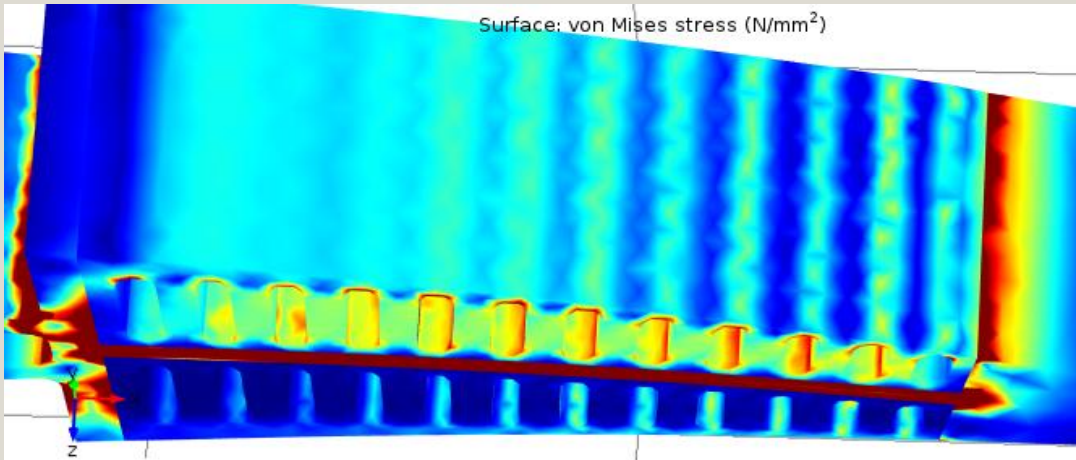
Lift change
integrated from
airfoil surface
pressure taps



NEXT GENERATION DESIGN (1)

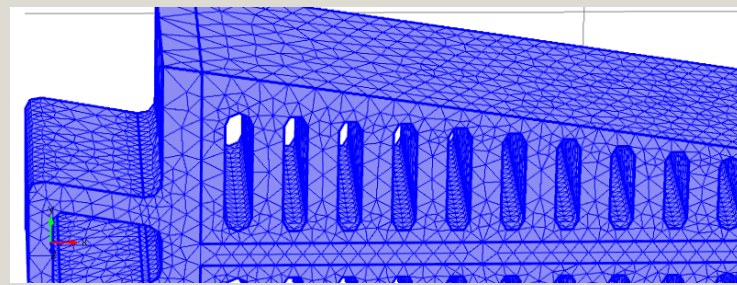


Original design



Next generation

NEXT GENERATION DESIGN (2)



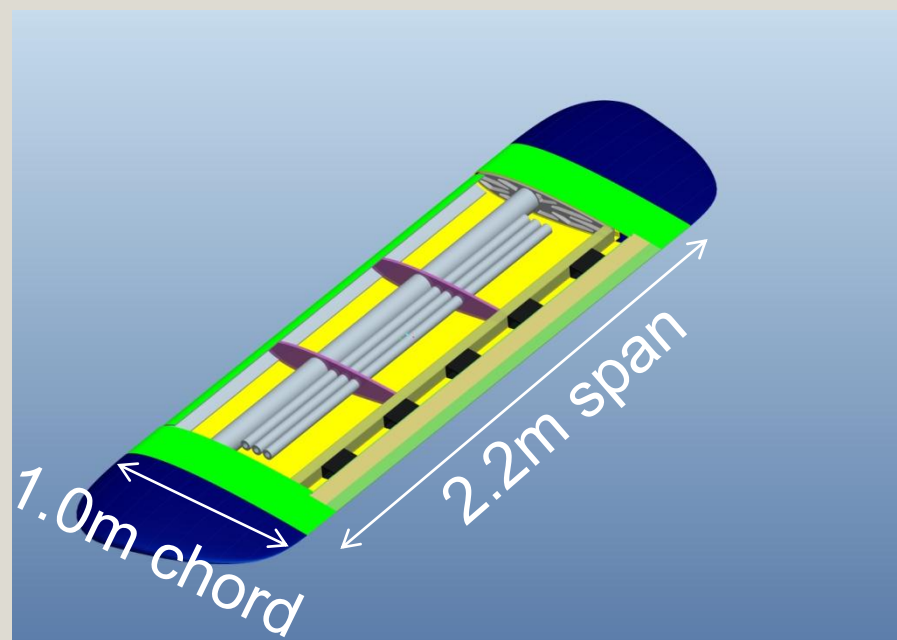
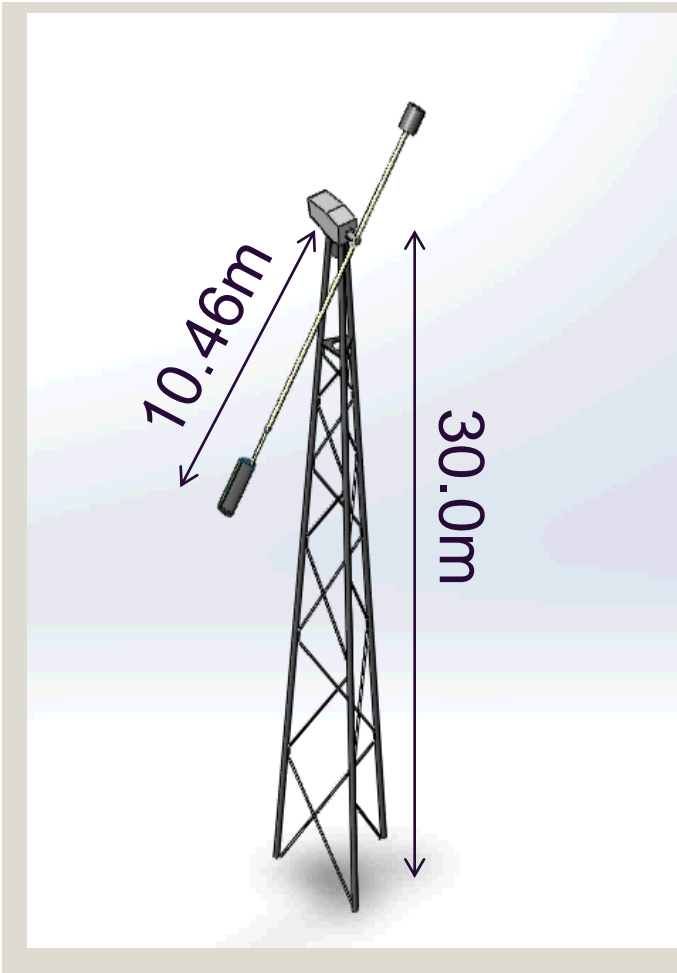
IN COLLABORATION WITH REHAU

Challenges in the implementation of the flap system on MW turbines

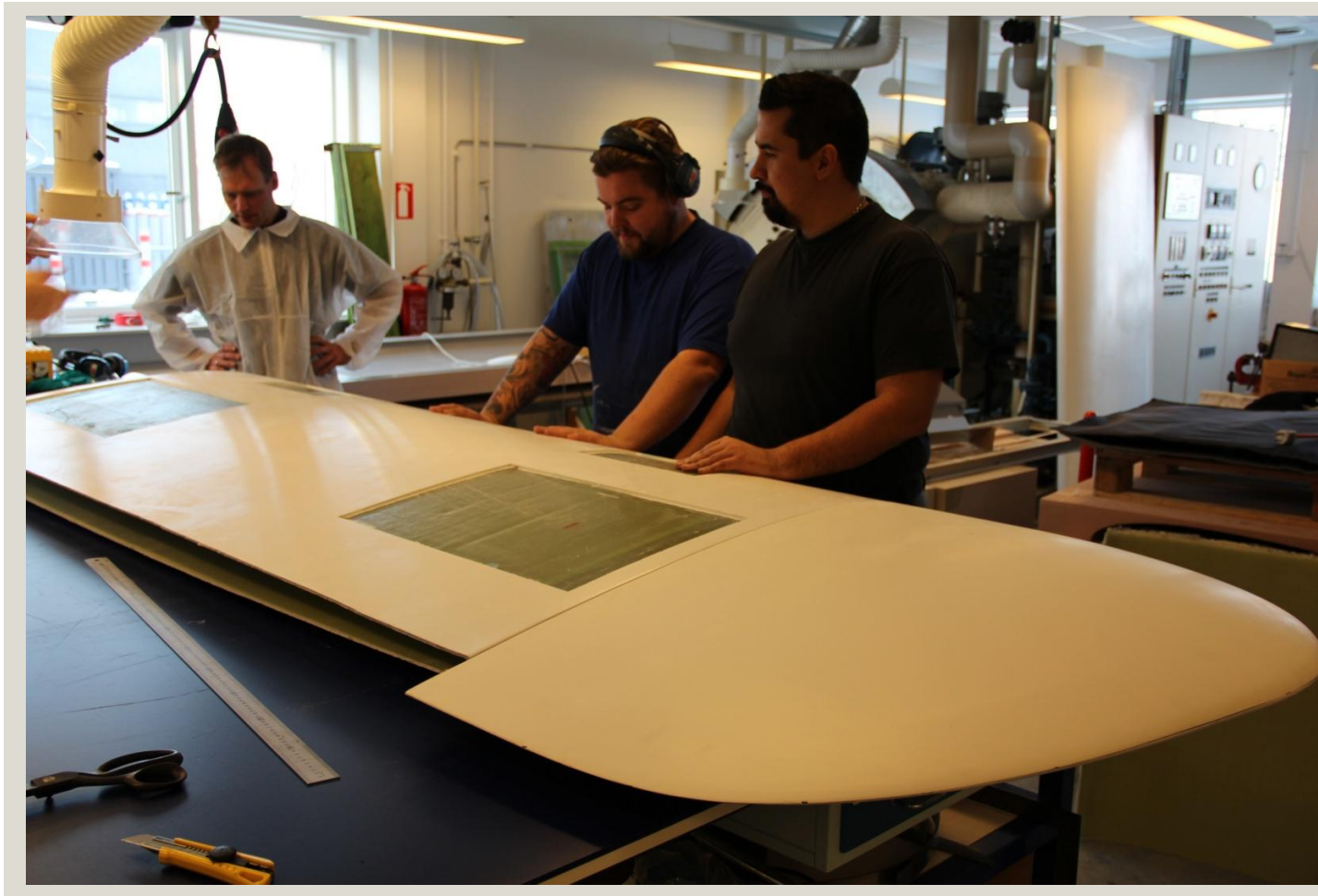
A NOVEL ROTATING TEST RIG (1)



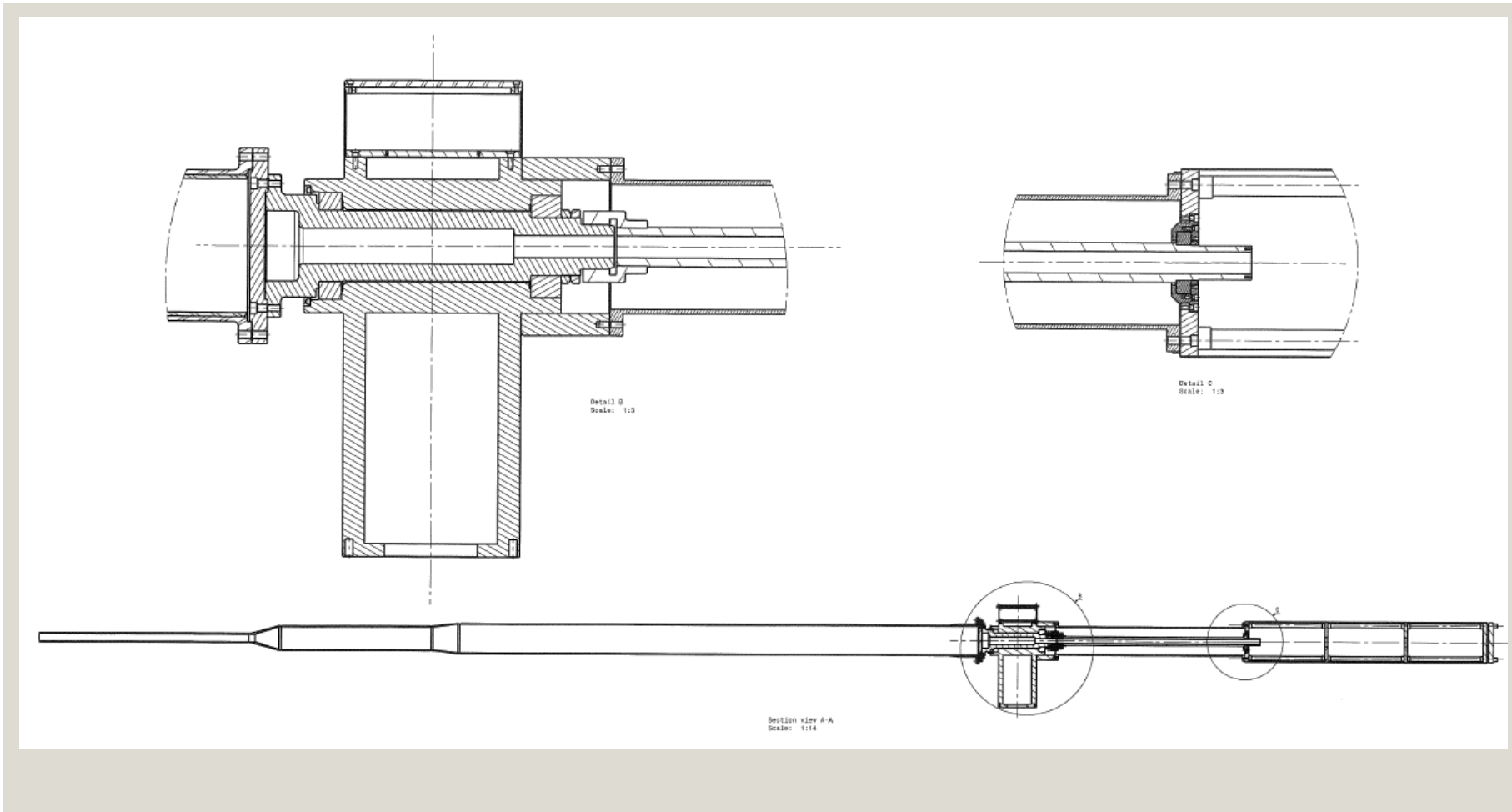
A NOVEL ROTATING TEST RIG (2)



A NOVEL ROTATING TEST RIG (3)

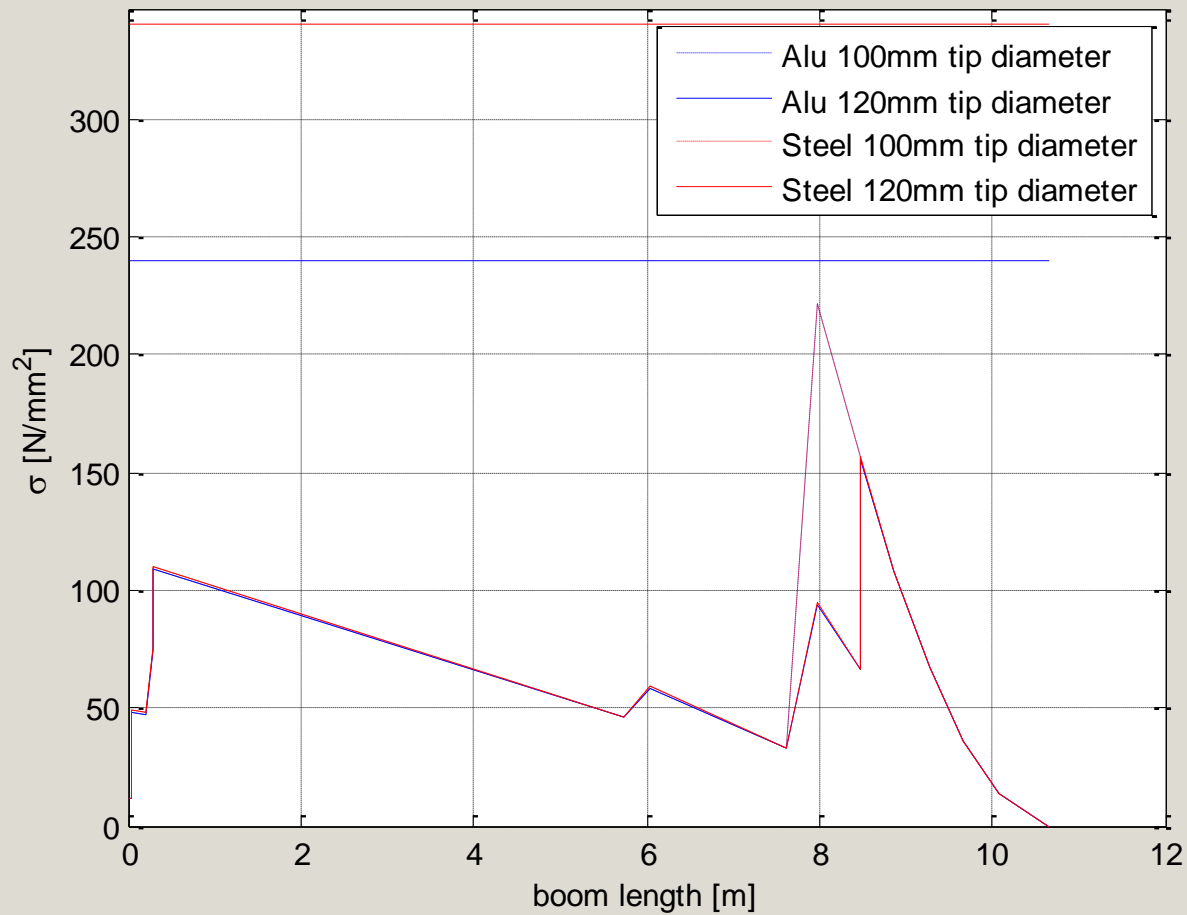


BOOM DESIGN (1)

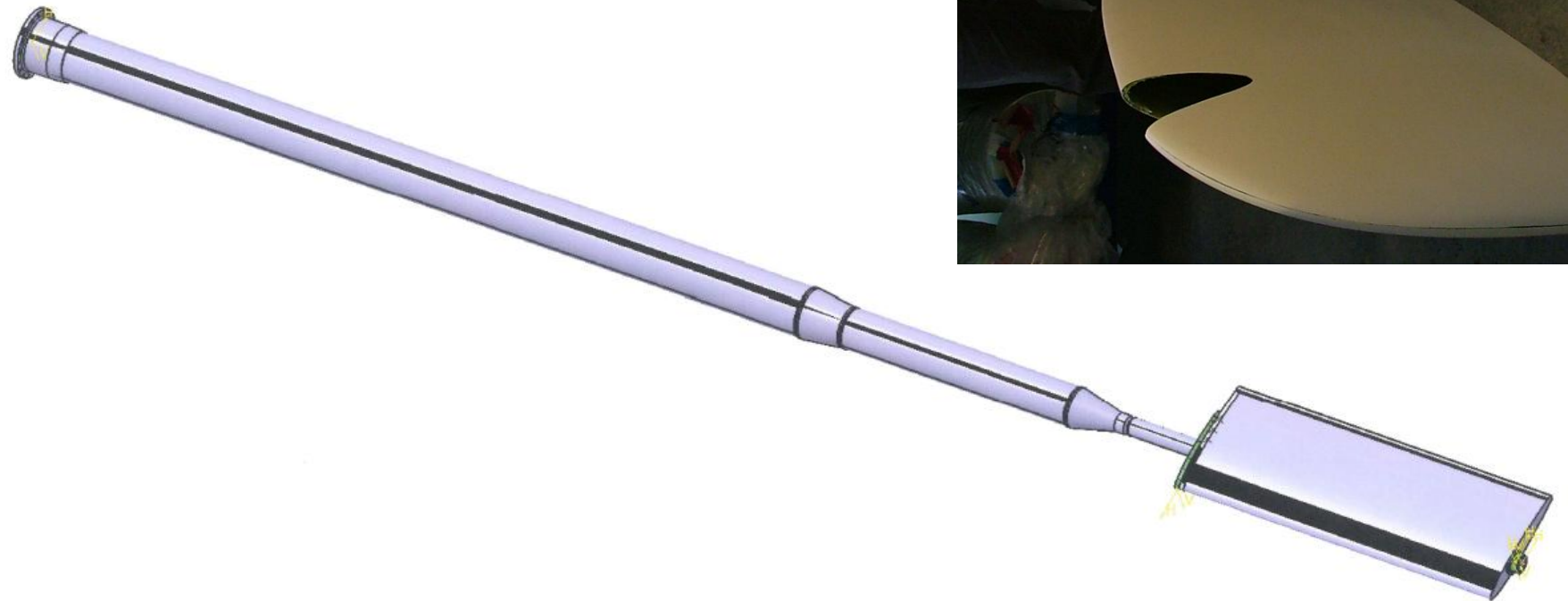


BOOM DESIGN (2)

Steady state stress

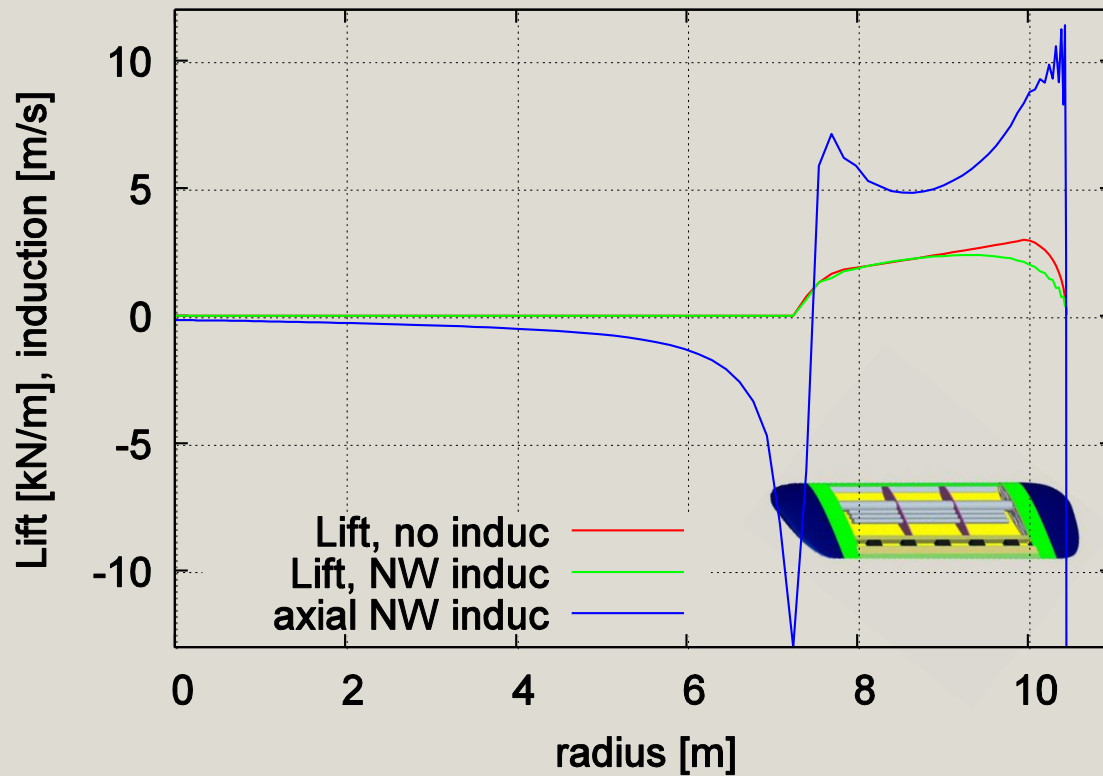


AERODYNAMICS (1)



AERODYNAMICS (2)

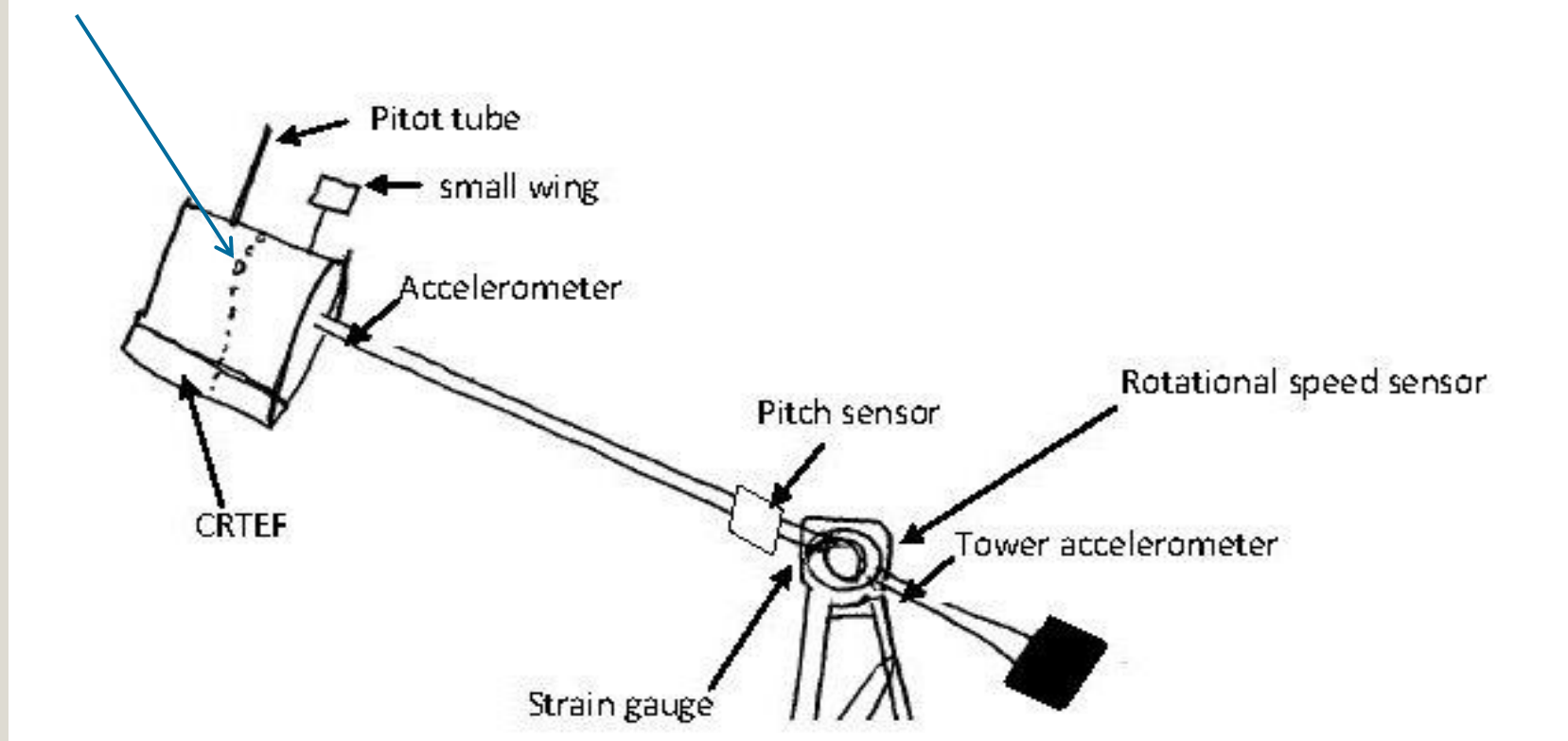
50 rpm, 8 degree pitch



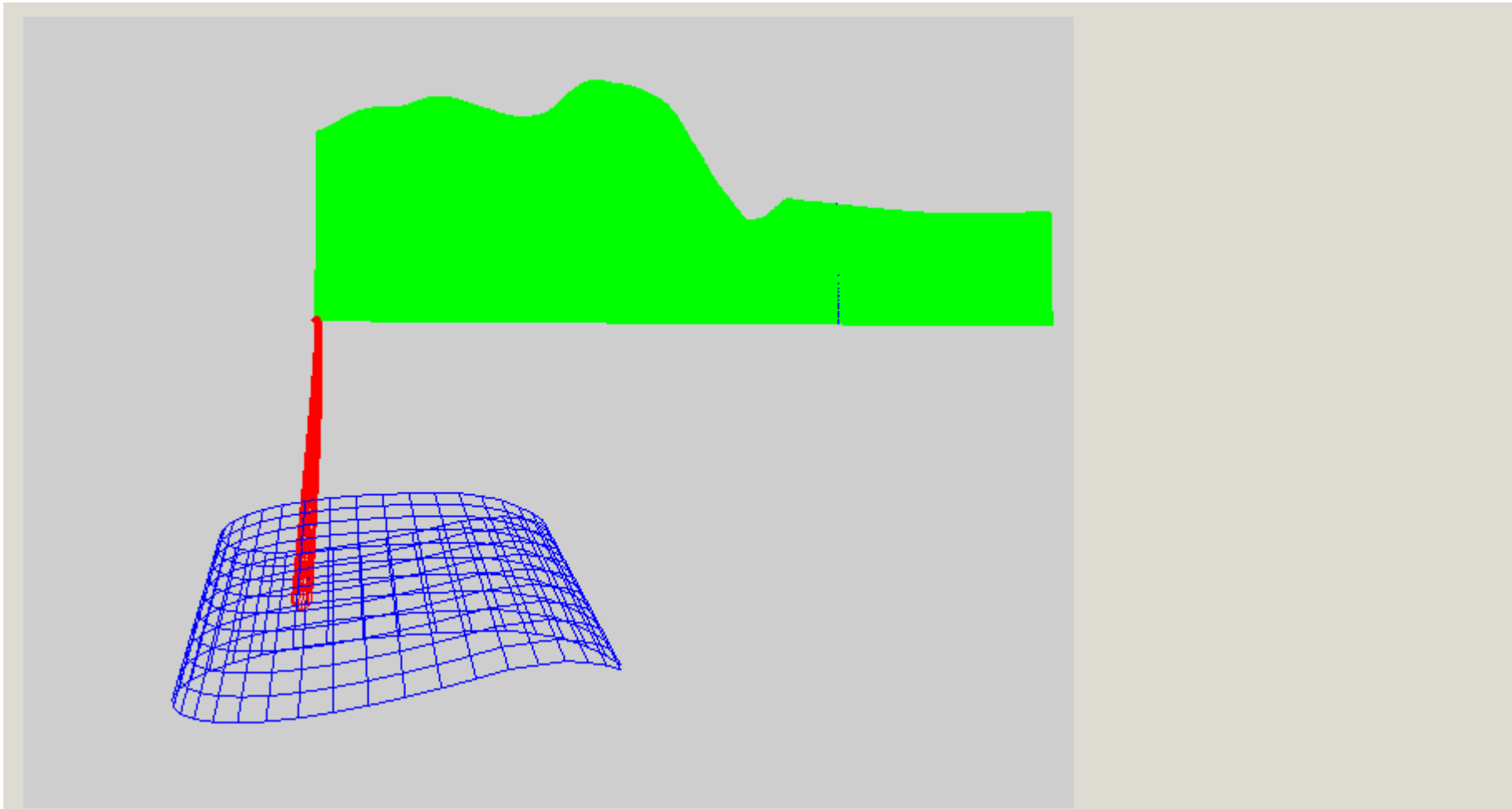
Courtesy of Georg Pirrung

AERODYNAMICS (3)

Pressure measurements



WHAT WE HOPE TO SEE...



OUTLOOK

- tests of flap system and control procedures on the rotating rig in summer 2013
- fatigue tests of flap system in laboratory autumn 2013
- late 2013 evaluation of the flap system for a full scale turbine

ACKNOWLEDGEMENT

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- Dansk Gummi Industri
- DTU (Elektro, Fiberlab, AED)